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NEWS 7 DEC 12

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TERMINAL (ENTER 1, 2, 3, OR ?):2

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NEWS 6 DEC 01 ChemPort single article sales feature unavailable

GBFULL now offers single source for full-text

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NEWS 8 DEC 17 Fifty-one pharmaceutical ingredients added to PS
NEWS 9 JAN 06 The retention policy for unread STNmail messages
will change in 2009 for STN-Columbus and STN-Tokyo

NEWS 10 JAN 07 WPIDS, WPINDEX, and WPIX enhanced Japanese Patent Classification Data

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FILE 'HOME' ENTERED AT 19:41:09 ON 31 JAN 2009

=> file registry COST IN U.S. DOLLARS

SINCE FILE TOTAL ENTRY SESSION 0.22 0.22

FULL ESTIMATED COST

FILE 'REGISTRY' ENTERED AT 19:41:54 ON 31 JAN 2009
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STRUCTURE FILE UPDATES: 29 JAN 2009 HIGHEST RN 1097778-52-3 DICTIONARY FILE UPDATES: 29 JAN 2009 HIGHEST RN 1097778-52-3

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http://www.cas.org/support/stngen/stndoc/properties.html

=> s 0.1-0.3 C/mac and 1.0-1.5 Mn/mac and 0.8-1.2 Si/mac and 1.0-1.4 Cr/mac and 0.1-0.3 Mo/mac 475538 0.1-0.3/MAC 273599 C/MAC

145576 0.1-0.3 C/MAC (P) C/MAC) 358258 1.0-1.5/MAC

393253 MN/MAC

102131 1.0-1.5 MN/MAC (1.0-1.5/MAC (P) MN/MAC)

356971 0.8-1.2/MAC

420616 SI/MAC 68891 0.8-1.2 SI/MAC

(0.8-1.2/MAC (P) SI/MAC)

323205 1.0-1.4/MAC 345555 CR/MAC

32082 1.0-1.4 CR/MAC (1.0-1.4/MAC (P) CR/MAC)

475538 0.1-0.3/MAC 207475 MO/MAC

53856 0.1-0.3 MO/MAC (0.1-0.3/MAC (P) MO/MAC)

L1 1415 0.1-0.3 C/MAC AND 1.0-1.5 MN/MAC AND 0.8-1.2 SI/MAC AND 1.0-1.4 CR/MAC AND 0.1-0.3 MO/MAC

=> 11 and 0-0.30 Cu/mac

L1 IS NOT A RECOGNIZED COMMAND

The previous command name entered was not recognized by the system. For a list of commands available to you in the current file, enter

"HELP COMMANDS" at an arrow prompt (=>).

=> s 11 and 0-0.30 Cu/mac 475715 0-0.30/MAC 245265 CU/MAC 60826 0-0.30 CU/MAC

(0-0.30/MAC (P) CU/MAC)

L2 692 L1 AND 0-0.30 CU/MAC

=> file caplus, uspatfull, scisearch COST IN U.S. DOLLARS

SINCE FILE TOTAL ENTRY SESSION FULL ESTIMATED COST 33.54 33.76

FILE 'CAPLUS' ENTERED AT 19:43:41 ON 31 JAN 2009 USE IS SUBJECT TO THE TERMS OF YOUR STN CUSTOMER AGREEMENT. PLEASE SEE "HELP USAGETERMS" FOR DETAILS. COPYRIGHT (C) 2009 AMERICAN CHEMICAL SOCIETY (ACS) FILE 'USPATFULL' ENTERED AT 19:43:41 ON 31 JAN 2009 CA INDEXING COPYRIGHT (C) 2009 AMERICAN CHEMICAL SOCIETY (ACS) FILE 'SCISEARCH' ENTERED AT 19:43:41 ON 31 JAN 2009 Copyright (c) 2009 The Thomson Corporation => s l1 and steel 'MAC' IS NOT A VALID FIELD CODE 1668 L1 AND STEEL => s 12 and steel 'MAC' IS NOT A VALID FIELD CODE 816 L2 AND STEEL => s 14 and patent/dt 718 L4 AND PATENT/DT => d 15 718 ANSWER 718 OF 718 USPATFULL on STN 80:44867 USPATFULL AN ΤI Method of manufacturing rolled wire rod IN Theis, Klaus, Hamburg, Germany, Federal Republic of Forster, Eckehard, Seevetal, Germany, Federal Republic of Becher, Wolfram, Neumunster, Germany, Federal Republic of Topfer, deceased, Hans-Jorg, late of Buxtehude, Germany, Federal Republic of by Marion o pfer, heiress PA Hamburger Stahlwerke GmbH, Hamburg, Germany, Federal Republic of (non-U.S. corporation) ΡI US 4222257 19800916 ΑI US 1977-848911 19771107 (5) PRAI DE 1977-2717780 19770421 DT Utility FS Granted LN.CNT 349 TNCI. INCLM: 072/201.000 INCLS: 072/700.000: 148/012.000B NCL NCLM: 072/201.000 NCLS: 072/700.000; 148/596.000; 148/664.000 TC [1] ICM B21B045-02 ICS C21D008-06: C21D009-52 IPCI B21B0045-02 [ICM,1]; C21D0008-06 [ICS,1]; C21D0009-52 [ICS,1] B21B0039-20 [I,C*]; B21B0039-28 [I,A]; B21B0045-02 [I,C*]; IPCR B21B0045-02 [I,A]; C21D0008-06 [I,C*]; C21D0008-06 [I,A]; C22C0038-02 [I,C*]; C22C0038-02 [I,A]; C22C0038-04 [I,C*]; C22C0038-04 [I,A] 072/201; 072/364; 072/365; 072/700; 148/12B; 148/12.4; 148/156 CAS INDEXING IS AVAILABLE FOR THIS PATENT.

=> d 15 718 abs

L5 ANSWER 718 OF 718 USPATFULL on STN

AB A method and apparatus of manufacturing wire rod having a content of

silicon and manganese greater than 1.5% is described, wherein the average final rolling temperature in processing the rod is from 870° C. to 970° C. and the rod is cooled in an extended position. The composition of steels utilized in the method is also described.

CAS INDEXING IS AVAILABLE FOR THIS PATENT.

=> file registry

FILE 'REGISTRY' ENTERED AT 19:45:47 ON 31 JAN 2009 USE IS SUBJECT TO THE TERMS OF YOUR SIN CUSTOMER AGREEMENT. PLEASE SEE "HELP USAGETERMS" FOR DETAILS.

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http://www.cas.org/support/stngen/stndoc/properties.html

=> d 12 abs

'ABS' IS NOT A VALID FORMAT FOR FILE 'REGISTRY'

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REG - RN

SAM - Index Name, MF, and structure - no RN

FIDE - All substance data, except sequence data

IDE - FIDE, but only 50 names

SQIDE - IDE, plus sequence data

SQIDE3 - Same as SQIDE, but 3-letter amino acid codes are used

SQD - Protein sequence data, includes RN

SQD3 - Same as SQD, but 3-letter amino acid codes are used SQN - Protein sequence name information, includes RN

EPROP - Table of experimental properties

PPROP - Table of predicted properties PROP - EPROP, ETAG, PPROP and SPEC Any CA File format may be combined with any substance format to obtain CA references citing the substance. The substance formats must be cited first. The CA File predefined formats are:

ABS -- Abstract

APPS -- Application and Priority Information

BIB -- CA Accession Number, plus Bibliographic Data

CAN -- CA Accession Number

CBIB -- CA Accession Number, plus Bibliographic Data (compressed)

IND -- Index Data

IPC -- International Patent Classification

PATS -- PI, SO

STD -- BIB, IPC, and NCL

IABS -- ABS, indented, with text labels

IBIB -- BIB, indented, with text labels

ISTD -- STD format, indented

OBIB ----- AN, plus Bibliographic Data (original)

OIBIB ----- OBIB, indented with text labels

SBIB ----- BIB, no citations

SIBIB ----- IBIB, no citations

The ALL format gives FIDE BIB ABS IND RE, plus sequence data when it is available.

The MAX format is the same as ALL.

The IALL format is the same as ALL with BIB ABS and IND indented, with text labels.

For additional information, please consult the following help messages:

HELP FORMATS -- To see a complete list of individual display fields. HELP FORMATS -- To see detailed descriptions of the predefined formats. ENTER DISPLAY FORMAT (IDE):sbib

'SBIB' IS NOT A VALID FORMAT FOR FILE 'REGISTRY'

The following are valid formats:

Substance information can be displayed by requesting individual fields or predefined formats. The predefined substance formats are: (RN = CAS Registry Number)

REG - RN

SAM - Index Name, MF, and structure - no RN

FIDE - All substance data, except sequence data

IDE - FIDE, but only 50 names

SQIDE - IDE, plus sequence data

SQIDE3 - Same as SQIDE, but 3-letter amino acid codes are used

SQD - Protein sequence data, includes RN

SQD3 - Same as SQD, but 3-letter amino acid codes are used

SQN - Protein sequence name information, includes RN

EPROP - Table of experimental properties

PPROP - Table of predicted properties

PROP - EPROP, ETAG, PPROP and SPEC

Any CA File format may be combined with any substance format to obtain CA references citing the substance. The substance formats must be cited first. The CA File predefined formats are:

```
ABS -- Abstract
APPS -- Application and Priority Information
BIB -- CA Accession Number, plus Bibliographic Data
CAN -- CA Accession Number
CBIB -- CA Accession Number, plus Bibliographic Data (compressed)
IND -- Index Data
IPC -- International Patent Classification
PATS -- PI, SO
STD -- BIB, IPC, and NCL
IABS -- ABS, indented, with text labels
IBIB -- BIB, indented, with text labels
ISTD -- STD format, indented
OBIB ----- AN, plus Bibliographic Data (original)
OIBIB ----- OBIB, indented with text labels
SBIB ----- BIB, no citations
SIBIB ----- IBIB, no citations
The ALL format gives FIDE BIB ABS IND RE, plus sequence data when
it is available.
The MAX format is the same as ALL.
The IALL format is the same as ALL with BIB ABS and IND indented.
with text labels.
For additional information, please consult the following help
messages:
HELP DFIELDS -- To see a complete list of individual display fields.
HELP FORMATS -- To see detailed descriptions of the predefined formats.
ENTER DISPLAY FORMAT (IDE):cancel
'CANCEL' IS NOT A VALID FORMAT FOR FILE 'REGISTRY'
The following are valid formats:
Substance information can be displayed by requesting individual
fields or predefined formats. The predefined substance formats
are: (RN = CAS Registry Number)
REG
SAM
      - Index Name, MF, and structure - no RN
FIDE
      - All substance data, except sequence data
IDE
      - FIDE, but only 50 names
SQIDE - IDE, plus sequence data
SQIDE3 - Same as SQIDE, but 3-letter amino acid codes are used
      - Protein sequence data, includes RN
SOD
     - Same as SOD, but 3-letter amino acid codes are used
SQD3
SON
      - Protein sequence name information, includes RN
EPROP - Table of experimental properties
PPROP - Table of predicted properties
PROP - EPROP, ETAG, PPROP and SPEC
Any CA File format may be combined with any substance format to
obtain CA references citing the substance. The substance formats
must be cited first. The CA File predefined formats are:
```

ABS -- Abstract

CAN -- CA Accession Number

APPS -- Application and Priority Information BIB -- CA Accession Number, plus Bibliographic Data

```
CBIB -- CA Accession Number, plus Bibliographic Data (compressed)
IND -- Index Data
IPC -- International Patent Classification
PATS -- PI, SO
STD -- BIB, IPC, and NCL
IABS -- ABS, indented, with text labels
IBIB -- BIB, indented, with text labels
ISTD -- STD format, indented
OBIB ----- AN, plus Bibliographic Data (original)
OIBIB ----- OBIB, indented with text labels
SBIB ----- BIB, no citations
SIBIB ----- IBIB, no citations
The ALL format gives FIDE BIB ABS IND RE, plus sequence data when
it is available.
The MAX format is the same as ALL.
The IALL format is the same as ALL with BIB ABS and IND indented,
with text labels.
For additional information, please consult the following help
messages:
HELP DFIELDS -- To see a complete list of individual display fields.
HELP FORMATS -- To see detailed descriptions of the predefined formats.
ENTER DISPLAY FORMAT (IDE):exit
'EXIT' IS NOT A VALID FORMAT FOR FILE 'REGISTRY'
The following are valid formats:
Substance information can be displayed by requesting individual
fields or predefined formats. The predefined substance formats
are: (RN = CAS Registry Number)
REG
    - RN
SAM
      - Index Name, MF, and structure - no RN
FIDE - All substance data, except sequence data
      - FIDE, but only 50 names
SQIDE - IDE, plus sequence data
SQIDE3 - Same as SQIDE, but 3-letter amino acid codes are used
SOD
     - Protein sequence data, includes RN
SQD3 - Same as SQD, but 3-letter amino acid codes are used
SON
      - Protein sequence name information, includes RN
EPROP - Table of experimental properties
PPROP - Table of predicted properties
      - EPROP, ETAG, PPROP and SPEC
PROP
Any CA File format may be combined with any substance format to
obtain CA references citing the substance. The substance formats
must be cited first. The CA File predefined formats are:
ABS -- Abstract
APPS -- Application and Priority Information
BIB -- CA Accession Number, plus Bibliographic Data
CAN -- CA Accession Number
CBIB -- CA Accession Number, plus Bibliographic Data (compressed)
IND -- Index Data
IPC -- International Patent Classification
```

PATS -- PI, SO

```
STD -- BIB, IPC, and NCL
IABS -- ABS, indented, with text labels
IBIB -- BIB, indented, with text labels
ISTD -- STD format, indented
OBIB ----- AN, plus Bibliographic Data (original)
OIBIB ----- OBIB, indented with text labels
SBIB ----- BIB, no citations
SIBIB ----- IBIB, no citations
The ALL format gives FIDE BIB ABS IND RE, plus sequence data when
it is available.
The MAX format is the same as ALL.
The IALL format is the same as ALL with BIB ABS and IND indented,
with text labels.
For additional information, please consult the following help
messages:
HELP DFIELDS -- To see a complete list of individual display fields.
HELP FORMATS -- To see detailed descriptions of the predefined formats.
ENTER DISPLAY FORMAT (IDE):istd
'ISTD' IS NOT A VALID FORMAT FOR FILE 'REGISTRY'
The following are valid formats:
Substance information can be displayed by requesting individual
fields or predefined formats. The predefined substance formats
are: (RN = CAS Registry Number)
      - RN
REG
SAM
      - Index Name, MF, and structure - no RN
FIDE - All substance data, except sequence data
TDE
      - FIDE, but only 50 names
SQIDE - IDE, plus sequence data
SQIDE3 - Same as SQIDE, but 3-letter amino acid codes are used
SQD
      - Protein sequence data, includes RN
SQD3 - Same as SQD, but 3-letter amino acid codes are used
SON
     - Protein sequence name information, includes RN
EPROP - Table of experimental properties
PPROP - Table of predicted properties
PROP - EPROP, ETAG, PPROP and SPEC
Any CA File format may be combined with any substance format to
obtain CA references citing the substance. The substance formats
must be cited first. The CA File predefined formats are:
ABS -- Abstract
APPS -- Application and Priority Information
BIB -- CA Accession Number, plus Bibliographic Data
CAN -- CA Accession Number
CBIB -- CA Accession Number, plus Bibliographic Data (compressed)
IND -- Index Data
IPC -- International Patent Classification
PATS -- PI, SO
STD -- BIB, IPC, and NCL
IABS -- ABS, indented, with text labels
IBIB -- BIB, indented, with text labels
```

ISTD -- STD format, indented

OBIB ----- AN, plus Bibliographic Data (original) OIBIB ----- OBIB, indented with text labels

SBIB ----- BIB, no citations SIBIB ----- IBIB, no citations

The ALL format gives FIDE BIB ABS IND RE, plus sequence data when it is available.

The MAX format is the same as ALL.

The IALL format is the same as ALL with BIB ABS and IND indented, with text labels.

For additional information, please consult the following help messages:

HELP DFIELDS -- To see a complete list of individual display fields. HELP FORMATS -- To see detailed descriptions of the predefined formats. ENTER DISPLAY FORMAT (IDE):reg IN 1093218-16-6 REGISTRY

=> file registry COST IN U.S. DOLLARS

SINCE FILE TOTAL ENTRY SESSION 1.21 92.77

FULL ESTIMATED COST

FILE 'REGISTRY' ENTERED AT 19:46:44 ON 31 JAN 2009 USE IS SUBJECT TO THE TERMS OF YOUR STN CUSTOMER AGREEMENT. PLEASE SEE "HELP USAGETERMS" FOR DETAILS. COPYRIGHT (C) 2009 American Chemical Society (ACS)

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=> d 12 2

L2 ANSWER 2 OF 692 REGISTRY COPYRIGHT 2009 ACS on STN

RN 1093216-10-4 REGISTRY

ED Entered STN: 12 Jan 2009

CN INDEX NAME NOT YET ASSIGNED

MF C . Al . Bi . Co . Cr . Cu . Fe . Ge . Mn . Mo . Ni . P . Sb . Se . Si . Sn . Te . Unspecified . W

I AYS

SR CA LC STN Files: CA, CAPLUS

Component	Comp	oon	ent	Compo	nent
	Per	cce	nt	Registry	Number
+				H======	
Fe	48	-	98	7439	9-89-6
Cr	0	-	15	7440)-47-3
Cu	0	_	8	7440	0-50-8
Al	2.5	-	6	7429	9-90-5
Co	0	-	4	7440)-48-4
Mo	0	-	4	7439	9-98-7
W	0	_	4	7440)-33-7
Si	0	_	3.5	7440)-21-3
Mn	0	-	3	7439	9-96-5
Ni	0	_	2	7440	0-02-0
Ge	0	_	0.5	7440)-56-4
Sb	0	-	0.5	7440	0-36-0
Sn	0	_	0.5	7440)-31-5
P	0	_	0.3	7723	3-14-0
Se	0	_	0.3	7782	2-49-2
Te	0	_	0.3	1349	4-80-9
Bi	0	_	0.2	7440)-69-9
C	0	_	0.1	7440	0-44-0
Misch metal	0	-	0.1	8049	9-20-5

3 REFERENCES IN FILE CA (1907 TO DATE)
3 REFERENCES IN FILE CAPLUS (1907 TO DATE)

=> file registry COST IN U.S. DOLLARS FULL ESTIMATED COST

SINCE FILE TOTAL ENTRY SESSION 2.53 95.30

FILE 'REGISTRY' ENTERED AT 19:47:07 ON 31 JAN 2009 USE IS SUBJECT TO THE TERMS OF YOUR STN CUSTOMER AGREEMENT. PLEASE SEE "HELP USAGETERMS" FOR DETAILS.

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STRUCTURE FILE UPDATES: 29 JAN 2009 HIGHEST RN 1097778-52-3 DICTIONARY FILE UPDATES: 29 JAN 2009 HIGHEST RN 1097778-52-3

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http://www.cas.org/support/stngen/stndoc/properties.html

 \Rightarrow s 0.1-0.3 C/mac and 1.0-1.5 Mn/mac and 0.8-1.2 Si/mac and 1.0-1.4 Cr/mac and 0.1-0.3 Mo/mac and 0-0.3 Cu/mac and 0-0.1 Bi/mac and 0-0.2 Pb/mac and 0-0.3 ni/mac and 0-0.1 S/mac and 80-100 Fe/mac

```
475538 0.1-0.3/MAC
        273599 C/MAC
        145576 0.1-0.3 C/MAC
                 (0.1-0.3/MAC (P) C/MAC)
       358258 1.0-1.5/MAC
       393253 MN/MAC
        102131 1.0-1.5 MN/MAC
                (1.0-1.5/MAC (P) MN/MAC)
       356971 0.8-1.2/MAC
        420616 SI/MAC
        68891 0.8-1.2 SI/MAC
                 (0.8-1.2/MAC (P) SI/MAC)
       323205 1.0-1.4/MAC
       345555 CR/MAC
        32082 1.0-1.4 CR/MAC
                (1.0-1.4/MAC (P) CR/MAC)
        475538 0.1-0.3/MAC
        207475 MO/MAC
        53856 0.1-0.3 MO/MAC
                 (0.1-0.3/MAC (P) MO/MAC)
        475715 0-0.3/MAC
        245265 CU/MAC
        60826 0-0.3 CU/MAC
                 (0-0.3/MAC (P) CU/MAC)
        336565 0-0.1/MAC
         20543 BT/MAC
         3203 0-0.1 BT/MAC
                 (0-0.1/MAC (P) BI/MAC)
        434955 0-0.2/MAC
        25740 PB/MAC
         6327 0-0.2 PB/MAC
                 (0-0.2/MAC (P) PB/MAC)
        475715 0-0.3/MAC
        341765 NI/MAC
        53982 0-0.3 NI/MAC
                 (0-0.3/MAC (P) NI/MAC)
       336565 0-0.1/MAC
         26299 S/MAC
        17353 0-0.1 S/MAC
                 (0-0.1/MAC (P) S/MAC)
        524182 80-100/MAC
        558510 FE/MAC
        272393 80-100 FE/MAC
                 (80-100/MAC (P) FE/MAC)
            19 0.1-0.3 C/MAC AND 1.0-1.5 MN/MAC AND 0.8-1.2 SI/MAC AND 1.0-1.4
               CR/MAC AND 0.1-0.3 MO/MAC AND 0-0.3 CU/MAC AND 0-0.1 BI/MAC AND
               0-0.2 PB/MAC AND 0-0.3 NI/MAC AND 0-0.1 S/MAC AND 80-100 FE/MAC
=> s 0.1-0.3 C/mac and 1.0-1.5 Mn/mac and 0.8-1.2 Si/mac and 1.0-1.4 Cr/mac and
0.1-0.3 Mo/mac and 0-0.3 Cu/mac and 0-0.1 Bi/mac and 0-0.2 Pb/mac and 0-0.3 ni/mac
and 0-0.1 S/mac and 80-95 Fe/mac
        475538 0.1-0.3/MAC
        273599 C/MAC
        145576 0.1-0.3 C/MAC
                 (0.1-0.3/MAC (P) C/MAC)
       358258 1.0-1.5/MAC
        393253 MN/MAC
        102131 1.0-1.5 MN/MAC
                (1.0-1.5/MAC (P) MN/MAC)
       356971 0.8-1.2/MAC
        420616 SI/MAC
        68891 0.8-1.2 SI/MAC
```

L6

```
(0.8-1.2/MAC (P) SI/MAC)
        323205 1.0-1.4/MAC
        345555 CR/MAC
        32082 1.0-1.4 CR/MAC
                (1.0-1.4/MAC (P) CR/MAC)
        475538 0.1-0.3/MAC
        207475 MO/MAC
         53856 0.1-0.3 MO/MAC
                (0.1-0.3/MAC (P) MO/MAC)
        475715 0-0.3/MAC
        245265 CU/MAC
        60826 0-0.3 CU/MAC
                (0-0.3/MAC (P) CU/MAC)
        336565 0-0.1/MAC
        20543 BT/MAC
         3203 0-0.1 BI/MAC
                (0-0.1/MAC (P) BI/MAC)
        434955 0-0.2/MAC
        25740 PB/MAC
         6327 0-0.2 PB/MAC
                 (0-0.2/MAC (P) PB/MAC)
        475715 0-0.3/MAC
        341765 NI/MAC
        53982 0-0.3 NI/MAC
                 (0-0.3/MAC (P) NI/MAC)
        336565 0-0.1/MAC
         26299 S/MAC
         17353 0-0.1 S/MAC
                 (0-0.1/MAC (P) S/MAC)
       318951 80-95/MAC
        558510 FE/MAC
       147519 80-95 FE/MAC
                (80-95/MAC (P) FE/MAC)
            19 0.1-0.3 C/MAC AND 1.0-1.5 MN/MAC AND 0.8-1.2 SI/MAC AND 1.0-1.4
               CR/MAC AND 0.1-0.3 MO/MAC AND 0-0.3 CU/MAC AND 0-0.1 BI/MAC AND
               0-0.2 PB/MAC AND 0-0.3 NI/MAC AND 0-0.1 S/MAC AND 80-95 FE/MAC
=> s 0.1-0.3 C/mac and 1.0-1.5 Mn/mac and 0.8-1.2 Si/mac and 1.0-1.4 Cr/mac and
0.1-0.3 Mo/mac and 0-0.3 Cu/mac and 0-0.1 Bi/mac and 0-0.2 Pb/mac and 0-0.3 ni/mac
and 0-0.1 S/mac and 70-100 Fe/mac
        475538 0.1-0.3/MAC
        273599 C/MAC
       145576 0.1-0.3 C/MAC
                 (0.1-0.3/MAC (P) C/MAC)
       358258 1.0-1.5/MAC
       393253 MN/MAC
       102131 1.0-1.5 MN/MAC
                 (1.0-1.5/MAC (P) MN/MAC)
       356971 0.8-1.2/MAC
        420616 SI/MAC
        68891 0.8-1.2 SI/MAC
                 (0.8-1.2/MAC (P) SI/MAC)
        323205 1.0-1.4/MAC
        345555 CR/MAC
        32082 1.0-1.4 CR/MAC
                 (1.0-1.4/MAC (P) CR/MAC)
        475538 0.1-0.3/MAC
        207475 MO/MAC
         53856 0.1-0.3 MO/MAC
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        475715 0-0.3/MAC
        245265 CU/MAC
```

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60826 0-0.3 CU/MAC
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        336565 0-0.1/MAC
        20543 BI/MAC
         3203 0-0.1 BI/MAC
                 (0-0.1/MAC (P) BI/MAC)
        434955 0-0.2/MAC
         25740 PB/MAC
         6327 0-0.2 PB/MAC
                (0-0.2/MAC (P) PB/MAC)
        475715 0-0.3/MAC
       341765 NI/MAC
        53982 0-0.3 NI/MAC
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        26299 S/MAC
        17353 0-0.1 S/MAC
                 (0-0.1/MAC (P) S/MAC)
       639931 70-100/MAC
        558510 FE/MAC
        328593 70-100 FE/MAC
                 (70-100/MAC (P) FE/MAC)
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               CR/MAC AND 0.1-0.3 MO/MAC AND 0-0.3 CU/MAC AND 0-0.1 BI/MAC AND
               0-0.2 PB/MAC AND 0-0.3 NI/MAC AND 0-0.1 S/MAC AND 70-100 FE/MAC
=> s 0.1-0.3 C/mac and 1.0-1.5 Mn/mac and 0.8-1.2 Si/mac and 1.0-1.4 Cr/mac and
0.1-0.3 Mo/mac and 0-0.3 Cu/mac and 0-0.1 Bi/mac and 0-0.2 Pb/mac and 0-0.3 ni/mac
and 70-100 Fe/mac
        475538 0.1-0.3/MAC
        273599 C/MAC
       145576 0.1-0.3 C/MAC
                (0.1-0.3/MAC (P) C/MAC)
        358258 1.0-1.5/MAC
       393253 MN/MAC
       102131 1.0-1.5 MN/MAC
                (1.0-1.5/MAC (P) MN/MAC)
       356971 0.8-1.2/MAC
        420616 SI/MAC
        68891 0.8-1.2 SI/MAC
                 (0.8-1.2/MAC (P) SI/MAC)
       323205 1.0-1.4/MAC
       345555 CR/MAC
        32082 1.0-1.4 CR/MAC
                 (1.0-1.4/MAC (P) CR/MAC)
        475538 0.1-0.3/MAC
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         53856 0.1-0.3 MO/MAC
                 (0.1-0.3/MAC (P) MO/MAC)
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        245265 CU/MAC
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                 (0-0.1/MAC (P) BI/MAC)
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         25740 PB/MAC
         6327 0-0.2 PB/MAC
                 (0-0.2/MAC (P) PB/MAC)
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341765 NT/MAC
        53982 0-0.3 NI/MAC
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                 (70-100/MAC (P) FE/MAC)
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=> s 0.1-0.3 C/mac and 1.0-1.5 Mn/mac and 0.8-1.2 Si/mac and 1.0-1.4 Cr/mac and
0.1-0.3 Mo/mac and 0-0.3 Cu/mac and 0-0.1 Bi/mac and 0-0.2 Pb/mac and 70-100 Fe/mac
        475538 0.1-0.3/MAC
        273599 C/MAC
       145576 0.1-0.3 C/MAC
                 (0.1-0.3/MAC (P) C/MAC)
        358258 1.0-1.5/MAC
        393253 MN/MAC
        102131 1.0-1.5 MN/MAC
                (1.0-1.5/MAC (P) MN/MAC)
        356971 0.8-1.2/MAC
        420616 SI/MAC
        68891 0.8-1.2 SI/MAC
                (0.8-1.2/MAC (P) SI/MAC)
        323205 1.0-1.4/MAC
        345555 CR/MAC
        32082 1.0-1.4 CR/MAC
                (1.0-1.4/MAC (P) CR/MAC)
        475538 0.1-0.3/MAC
        207475 MO/MAC
        53856 0.1-0.3 MO/MAC
                (0.1-0.3/MAC (P) MO/MAC)
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        245265 CU/MAC
        60826 0-0.3 CU/MAC
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         3203 0-0.1 BI/MAC
                 (0-0.1/MAC (P) BI/MAC)
        434955 0-0.2/MAC
         25740 PB/MAC
         6327 0-0.2 PB/MAC
                 (0-0.2/MAC (P) PB/MAC)
       639931 70-100/MAC
       558510 FE/MAC
       328593 70-100 FE/MAC
                 (70-100/MAC (P) FE/MAC)
            24 0.1-0.3 C/MAC AND 1.0-1.5 MN/MAC AND 0.8-1.2 SI/MAC AND 1.0-1.4
              CR/MAC AND 0.1-0.3 MO/MAC AND 0-0.3 CU/MAC AND 0-0.1 BI/MAC AND
               0-0.2 PB/MAC AND 70-100 FE/MAC
=> s 0.1-0.3 C/mac and 1.0-1.5 Mn/mac and 0.8-1.2 Si/mac and 1.0-1.4 Cr/mac and
0.1-0.3 Mo/mac and 0-0.3 Cu/mac and 0-0.1 Bi/mac and 70-100 Fe/mac
        475538 0.1-0.3/MAC
        273599 C/MAC
        145576 0.1-0.3 C/MAC
                 (0.1-0.3/MAC (P) C/MAC)
       358258 1.0-1.5/MAC
       393253 MN/MAC
       102131 1.0-1.5 MN/MAC
```

L9

L10

```
(1.0-1.5/MAC (P) MN/MAC)
        356971 0.8-1.2/MAC
        420616 SI/MAC
        68891 0.8-1.2 SI/MAC
                 (0.8-1.2/MAC (P) SI/MAC)
       323205 1.0-1.4/MAC
       345555 CR/MAC
        32082 1.0-1.4 CR/MAC
                (1.0-1.4/MAC (P) CR/MAC)
        475538 0.1-0.3/MAC
        207475 MO/MAC
        53856 0.1-0.3 MO/MAC
                (0.1-0.3/MAC (P) MO/MAC)
        475715 0-0.3/MAC
        245265 CU/MAC
        60826 0-0.3 CU/MAC
                (0-0.3/MAC (P) CU/MAC)
       336565 0-0.1/MAC
        20543 BI/MAC
         3203 0-0.1 BI/MAC
                 (0-0.1/MAC (P) BI/MAC)
       639931 70-100/MAC
        558510 FE/MAC
       328593 70-100 FE/MAC
                (70-100/MAC (P) FE/MAC)
            35 0.1-0.3 C/MAC AND 1.0-1.5 MN/MAC AND 0.8-1.2 SI/MAC AND 1.0-1.4
               CR/MAC AND 0.1-0.3 MO/MAC AND 0-0.3 CU/MAC AND 0-0.1 BI/MAC AND
               70-100 FE/MAC
=> s 0.1-0.3 C/mac and 1.0-1.5 Mn/mac and 0.8-1.2 Si/mac and 1.0-1.4 Cr/mac and
0.1-0.3 Mo/mac and 0-0.3 Cu/mac and 70-100 Fe/mac
        475538 0.1-0.3/MAC
        273599 C/MAC
        145576 0.1-0.3 C/MAC
                 (0.1-0.3/MAC (P) C/MAC)
       358258 1.0-1.5/MAC
       393253 MN/MAC
       102131 1.0-1.5 MN/MAC
                 (1.0-1.5/MAC (P) MN/MAC)
       356971 0.8-1.2/MAC
        420616 SI/MAC
        68891 0.8-1.2 SI/MAC
                 (0.8-1.2/MAC (P) SI/MAC)
       323205 1.0-1.4/MAC
       345555 CR/MAC
        32082 1.0-1.4 CR/MAC
                 (1.0-1.4/MAC (P) CR/MAC)
        475538 0.1-0.3/MAC
        207475 MO/MAC
         53856 0.1-0.3 MO/MAC
                 (0.1-0.3/MAC (P) MO/MAC)
        475715 0-0.3/MAC
        245265 CU/MAC
        60826 0-0.3 CU/MAC
                 (0-0.3/MAC (P) CU/MAC)
       639931 70-100/MAC
        558510 FE/MAC
       328593 70-100 FE/MAC
                 (70-100/MAC (P) FE/MAC)
          688 0.1-0.3 C/MAC AND 1.0-1.5 MN/MAC AND 0.8-1.2 SI/MAC AND 1.0-1.4
               CR/MAC AND 0.1-0.3 MO/MAC AND 0-0.3 CU/MAC AND 70-100 FE/MAC
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```
=> s 0.1-0.3 C/mac and 1.0-1.5 Mn/mac and 0.8-1.2 Si/mac and 1.0-1.4 Cr/mac and
0.1-0.3 Mo/mac and 0-0.3 Cu/mac and 0-0.1 Bi/mac and 70-100 Fe/mac
        475538 0.1-0.3/MAC
        273599 C/MAC
        145576 0.1-0.3 C/MAC
                (0.1-0.3/MAC (P) C/MAC)
        358258 1.0-1.5/MAC
        393253 MN/MAC
        102131 1.0-1.5 MN/MAC
                (1.0-1.5/MAC (P) MN/MAC)
        356971 0.8-1.2/MAC
        420616 SI/MAC
         68891 0.8-1.2 SI/MAC
                (0.8-1.2/MAC (P) SI/MAC)
        323205 1.0-1.4/MAC
        345555 CR/MAC
         32082 1.0-1.4 CR/MAC
                (1.0-1.4/MAC (P) CR/MAC)
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                (70-100/MAC (P) FE/MAC)
1.13
            35 0.1-0.3 C/MAC AND 1.0-1.5 MN/MAC AND 0.8-1.2 SI/MAC AND 1.0-1.4
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               70-100 FE/MAC
=> file caplus, uspatfull, scisearch
COST IN U.S. DOLLARS
                                                 SINCE FILE
                                                                TOTAL
                                                     ENTRY SESSION
FULL ESTIMATED COST
                                                     404.61
                                                              499.91
FILE 'CAPLUS' ENTERED AT 19:51:17 ON 31 JAN 2009
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FILE 'SCISEARCH' ENTERED AT 19:51:17 ON 31 JAN 2009
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=> s 113
'MAC' IS NOT A VALID FIELD CODE
L14
         43 L13
=> d 114 a11
L14 ANSWER 1 OF 43 CAPLUS COPYRIGHT 2009 ACS on STN
AN 2009:27896 CAPLUS
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Method for manufacture of nondirectional electromagnetic steel sheets for
IN Tanaka, Ichiro; Yashiki, Hiroyoshi; Takamaru, Hiroki; Fujita, Kaoru
PA
    Sumitomo Metal Industries Ltd., Japan
    Jpn. Kokai Tokkyo Koho, 19pp.
SO
    CODEN: JKXXAF
DT Patent
LA
   Japanese
CC
     55-11 (Ferrous Metals and Alloys)
FAN.CNT 1
     PATENT NO.
                        KIND
                                DATE
                                           APPLICATION NO.
                                                                 DATE
PI JP 2009001887
                         Α
                                20090108
                                           JP 2007-165962
                                                                  20070625
PRAI JP 2007-165962
                                20070625
CLASS
 PATENT NO.
                CLASS PATENT FAMILY CLASSIFICATION CODES
 JP 2009001887 IPCI
                        C21D0008-12 [I,A]; B21B0003-02 [I,A]; H01F0001-16
                        [I,A]; H01F0001-12 [I,C*]; H01F0041-02 [I,A];
                        C22C0038-00 [N,A]; C22C0038-14 [N,A]; C22C0038-60 [N,A]
                 FTERM 4K033/AA01; 4K033/CA01; 4K033/CA02; 4K033/CA03;
                        4K033/CA04; 4K033/CA05; 4K033/CA06; 4K033/CA07;
                        4K033/CA08; 4K033/CA09; 4K033/CA10; 4K033/EA02;
                        4K033/FA01; 4K033/FA03; 4K033/FA10; 4K033/FA12;
                        4K033/GA00; 4K033/HA01; 4K033/HA03; 4K033/JA01;
                        4K033/KA00; 5E041/AA02; 5E041/CA04; 5E041/HB11;
                        5E041/NN01; 5E062/AA06; 5E062/AC15
ΔR
    Slabs or ingots of steels consisting of C ≤0.06, Si ≤3.5, Mn
     0.05-3.0, Al >2.5 and \leq 6.0, P \leq 0.30, S \leq 0.04, N
     ≤0.02, Nb >0.02 weight%, and balance Fe and satisfying 0 <Nb/93 +
     Zr/91 + Ti/48 + V/51 - (C/12 + N/14) < 5 + 10-3 (the element symbols
     indicate their weight% contents) are hot rolled, cold rolled, at least once
     with in-between process annealing, into sheets of thickness 0.15-0.80 mm
     and tensile strength ≥850 MPa, and then soaked at
     ≤820° to give the title sheets. The steels may also contain
     Cu 0.01-8.0, Ni 0.01-2.0, Cr 0.01-15.0, Mo 0.005-4.0, Co 0.01-4.0, W
     0.01-4.0, Sn \leq0.5, Sb \leq0.5, Se \leq0.3, Bi \leq0.2,
     Ge \leq 0.5, Te \leq 0.3, B \leq 0.01, Ca \leq 0.03, Mg
     <0.02, and/or rare earth metals ≤0.1 weight%. Rotors with
     decreased weight can be prepared from the sheets.
ST
    nondirectional electromagnetic steel sheet rolling rotor
IT
     Rolling (metals)
        (hot; manufacture of nondirectional electromagnetic steel sheets for rotors
        by hot and cold rolling, process annealing, and soaking)
     Cold rolling
        (manufacture of nondirectional electromagnetic steel sheets for rotors by
        hot and cold rolling, process annealing, and soaking)
     Annealing
        (process; manufacture of nondirectional electromagnetic steel sheets for
        rotors by hot and cold rolling, process annealing, and soaking)
     Electric motors
        (rotors; manufacture of nondirectional electromagnetic steel sheets for
        rotors by hot and cold rolling, process annealing, and soaking)
     Heat treatment
        (soaking; manufacture of nondirectional electromagnetic steel sheets for
        rotors by hot and cold rolling, process annealing, and soaking)
     1093215-97-4 1093215-98-5 1093215-99-6 1093216-01-3
                                                                1093216-02-4
     1093216-04-6
                  1093216-05-7 1093216-06-8 1093216-07-9
     1093216-10-4
```

RL: PEP (Physical, engineering or chemical process); TEM (Technical or

DN

ED

150:81420

Entered STN: 09 Jan 2009

- engineered material use); PROC (Process); USES (Uses)
 - (manufacture of nondirectional electromagnetic steel sheets for rotors by hot and cold rolling, process annealing, and soaking)
- IT 7439-95-4, Magnesium, uses 7440-42-8, Boron, uses 7440-70-2, Calcium, uses
 - RL: MOA (Modifier or additive use); USES (Uses)

(microalloying element; manufacture of nondirectional electromagnetic steel sheets for rotors by hot and cold rolling, process annealing, and soaking)

=>	a	114	рi	1

L14	ANSWER 1 OF 43 PATENT NO.	CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO.	DATE
ΡI	JP 2009001887	A 20090108 JP 2007-165962	20070625
=> d	114 pi 1-43		
L14	ANSWER 1 OF 43 PATENT NO.	CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO.	DATE
PI		A 20090108 JP 2007-165962	
L14	ANSWER 2 OF 43 PATENT NO.	CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO. A 20090108 JP 2007-163851	DATE
PI	JP 2009001864	A 20090108 JP 2007-163851	20070621
L14	ANSWER 3 OF 43 PATENT NO.	CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO.	DATE
PI	JP 2008308727	A 20081225 JP 2007-157349	20070614
L14	ANSWER 4 OF 43 PATENT NO.	CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO. A 20081225 JP 2007-154899	DATE
PI	JP 2008308704	A 20081225 JP 2007-154899	20070612
L14	ANSWER 5 OF 43 PATENT NO.	CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO.	DATE
PI	JP 2008149340	A 20080703 JP 2006-338791	20061215
L14	ANSWER 6 OF 43 PATENT NO.	CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO.	DATE
PI	JP 2008013788	A 20080124 JP 2006-183512	20060703
L14	ANSWER 7 OF 43 PATENT NO.	CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO.	DATE
PI	JP 2007031787	A 20070208 JP 2005-217453	20050727
L14	ANSWER 8 OF 43 PATENT NO.	CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO.	DATE
PI	WO 2007007423 W: AE, AG,	A1 20070118 WO 2005-JP22368 AL, AM, AT, AU, AZ, BA, BB, BG, BR, BW, BY, CR, CU, CZ, DE, DK, DM, DZ, EC, EE, EG, ES,	20051206 BZ, CA, CH,

	LC, LK, NA, NG, SK, SL, YU, ZA, RW: AT, BE, IS, IT, CF, CG, GM, KE,	GM, HR, HU, ID, IL, IN, IS, KE, KG, KM, KN, LR, LS, LT, LU, LV, LY, MA, MD, MG, MK, MN, NI, NO, NZ, OM, PG, PH, PL, PT, RO, RU, SC, SM, SY, TJ, TM, TN, TR, TT, TZ, UA, UG, UG, SM, GL, MT, SM, SM, SM, SM, SM, SM, SM, SM, SM, SM	MW, MX, MZ, SD, SE, SG, UZ, VC, VN, GR, HU, IE, TR, BF, BJ, TG, BW, GH, AM, AZ, BY,
L14	PATENT NO.	CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO.	DATE
PI	JP 2006348321	A 20061228 JP 2005-173112	20050614
L14	ANSWER 10 OF 43 PATENT NO.	CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO. A 20061102 JP 2005-118366	DATE
PI	JP 2006299296	A 20061102 JP 2005-118366	20050415
L14	ANSWER 11 OF 43 PATENT NO.	CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO. A 20060921 JP 2005-64316	DATE
PI	JP 2006249457	A 20060921 JP 2005-64316	20050308
L14	ANSWER 12 OF 43 PATENT NO.	CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO. A 20060810 JP 2005-21019	DATE
PI	JP 2006206967	A 20060810 JP 2005-21019	20050128
L14	ANSWER 13 OF 43 PATENT NO.	CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO.	DATE
PI	JP 2006037177		20040728
L14	ANSWER 14 OF 43 PATENT NO.	CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO.	DATE
PI	FR 2868083 FR 2868083	A1 20050930 FR 2004-3038 B1 20060721	20040324
	AU 2005232002 CA 2559562 WO 2005098070 WO 2005098070	CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO. A1 20050930 FR 2004-3038 B1 20050931 A1 20051020 AU 2005-232002 A1 20051020 WO 2005-2559562 A2 20051020 WO 2005-PF6684 A3 20061005 AL AM AT AU AZ RA BB BC BB BW BY	20050321 20050321 20050321
	CN, CO, GE, GH, LK, LR, NO, NZ, SY, TJ, RW: BW, GH, AZ, BY, EE, ES, RO, SE,	AL, AM, AT, AU, AZ, BA, BB, BG, BR, BW, BY, CR, CU, CZ, DE, DK, DM, DZ, EC, EE, EG, ES, GM, HE, HU, ID, IL, IN, IS, JP, KE, KG, KP, LS, LT, LU, LV, MA, MD, MG, MK, MN, MW, MM, CM, PG, PH, PL, PT, RO, RU, SC, SD, SE, SG, TM, TM, TM, TR, TT, TZ, UA, UG, UZ, VC, VG, MG, KE, LS, MW, MZ, NA, SD, SI, SZ, TZ, UG, KG, KZ, MD, RU, TJ, TM, AT, BE, BG, CH, CY, FI, FF, GB, GR, HU, IE, IS, IT, LT, LU, MC, SI, SI, SK, TR, BF, BJ, CF, CG, CI, CM, GA, GN, SN, TD, TG	FI, GB, GD, KR, KZ, LC, MZ, NA, NI, SK, SL, SM, YU, ZA, ZM, ZW, ZM, ZW, AM, CZ, DE, DK, NL, PL, PT,

	IS, IT,	BG A2 20061206 EP 2005-742733 BG, CH, CY, CZ, DE, DK, EE, ES, FI, FR, GB, LI, LT, LU, MC, NL, PL, PT, RO, SE, SI, SK, MK, YU A 20070904 BR 2005-8776 T 200771101 JP 2007-504447 A1 20070823 US 2006-593463 A 20061120 MX 2006-10816 A 20070615 IN 2006-CN3491 A 20070222 KR 2006-722040	TR, AL, BA,
			20061024
L14	ANSWER 15 OF 43 PATENT NO.	CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO. A 20050818 JP 2004-30993	DATE
PI	JP 2005220423	A 20050818 JP 2004-30993	20040206
L14	ANSWER 16 OF 43 PATENT NO.	CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO. A 20050310 JP 2004-199624	DATE
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L14	ANSWER 17 OF 43 PATENT NO.	CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO. A 20041021 JP 2003-89959 B2 20070509	DATE
PI	JP 2004292929 JP 3912308	A 20041021 JP 2003-89959 B2 20070509	20030328
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PI		CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO. A 20040129 JP 2002-182228 B2 20060628	
L14	ANSWER 19 OF 43 PATENT NO.	CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO. A1 20030813 EP 2003-250686 B1 20050420	DATE
PI	EP 1335035 EP 1335035	A1 20030813 EP 2003-250686 B1 20050420	20030203
		CH, DE, DK, ES, PR, GB, GR, IT, LI, LU, NL, LT, LV, FI, RO, MK, CY, AL, TR, BG, CZ, EE, A 20030815 JP 2002-26368 B2 20060322 TW 2003-92102006 A 20030820 CN 2003-103453 CC 20050713 A1 20030814 US 2003-357419	
	TW 228149 CN 1436875 CN 1210432	B 20050221 TW 2003-92102006 A 20030820 CN 2003-103453 C 20050713	20030129 20030130
	US 20030152476	A1 20030814 US 2003-357419	20030204
L14	ANSWER 20 OF 43 PATENT NO.	CAPLUS COEYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO. A 20030521 JP 2001-349097 B2 20061129	DATE
PI	JP 2003147481 JP 3851146	A 20030521 JP 2001-349097 B2 20061129	20011114
L14	ANSWER 21 OF 43 PATENT NO.	CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO.	DATE
ΡI	JP 2003147479 JP 3901994	CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO.	20011114
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14 ANSWER 30 OF 43 CAPLUS COPYRIGHT 2009 ACS ON STN PATENT NO. EXIND DATE APPLICATION NO. DATE	PI	JP 2003073735 JP 4213370	A 200300 B2 200900	312 JP 2001-258353 121	20010828
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PI JP 2001131684 A 20010515 JP 1999-314366 19991104	L14	ANSWER 30 OF 43 PATENT NO.	CAPLUS COPYRIGI KIND DATE	HT 2009 ACS on STN APPLICATION NO.	DATE
L14 ANSWER 31 OF 43 CAPLUS COPYRIGHT 2009 ACS ON STN APPLICATION NO. DATE PI JP 2000273574 A 20001003 JP 1999-81629 19990325	PI	JP 2001131684	A 20010	515 JP 1999-314366	19991104
PI JP 2000273574 A 20001003 JP 1999-81629 19990325	L14	ANSWER 31 OF 43 PATENT NO.	CAPLUS COPYRIGI KIND DATE	HT 2009 ACS on STN APPLICATION NO.	DATE
	PI	JP 2000273574	A 200010	003 JP 1999-81629	19990325

L14	ANSWER 32 OF 43 PATENT NO.	CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO.	DATE
PI	JP 2000219936	A 20000808 JP 1999-23459	19990201
L14	ANSWER 33 OF 43 PATENT NO.	CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO	DATE
PI	WO 2000044953 W: CA, CN,	A1 20000803 WO 2000-JP369	20000125
		CH, CY, DE, DK, ES, FI, FR, GB, GR, IE, II	, LU, MC, NL,
	CA 2323952	A1 20000803 CA 2000-2323952 A1 20010117 EP 2000-900930	20000125
	CN 1113973	CH, DE, DK, ES, FR, GB, GR, IT, LI, LU, NI C 20030709 CN 2000-8000083 A 20001010 JP 2000-18101 B2 20050810 A 20010807 JP 2000-204882 B2 20050810 B1 20021105 US 2000-669552 CAPLUS COPYRIGHT 2009 ACS on STN	20000125
	JP 2000282172 JP 3680674	A 20001010 JP 2000-18101 B2 20050810	20000127
	JP 2001214241	A 20010807 JP 2000-204882	20000706
	US 6475305	B1 20021105 US 2000-669552	20000926
T 1 /	MCMED 34 OF 43	CAPLUS COPYRIGHT 2009 ACS on STN	
714	PATENT NO.	KIND DATE APPLICATION NO. A 20000125 JP 1998-192878 B2 20070314	DATE
PI	JP 2000026933	A 20000125 JP 1998-192878	19980708
	JP 3893756	B2 20070314	
L14	ANSWER 35 OF 43	CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO. A 19990106 JP 1998-52103 B2 20040126	DATE
PΙ	JP 3489656	B2 20040126 JP 1998-52103	19980304
	PATENT NO.	KIND DATE APPLICATION NO.	DATE
PI	JP 10152754	CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO. A 19980609 JP 1997-249242 B2 20051207	19970912
L14	ANSWER 37 OF 43	CAPLUS COPYRIGHT 2009 ACS on STN	DATE
	THIENT NO.	KIND DATE APPLICATION NO. A 19970128 JP 1995-175318	
PI			19950712
L14	ANSWER 38 OF 43	CAPLUS COPYRIGHT 2009 ACS on STN	DATE
	FAIENI NO.	KIND DATE APPLICATION NO. A 19950613 JP 1993-296667	
PI			19931126
L14	PATENT NO.	CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO.	DATE
PI	JP 06248341	A 19940906 JP 1993-33687	19930223
		USPATFULL on STN A1 20070823	
	ANSWER 41 OF 43 US 20030152476	USPATFULL on STN A1 20030814	

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L14 ANSWER 42 OF 43 USPATFULL on STN
PI US 20030084965 A1 20030508
US 6838048 B2 20050104
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L14 ANSWER 43 OF 43 USPATFULL on STN PI US 6475305 B1 20021105

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L22 ANSWER 1 OF 30 USPATFULL on STN
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AN 2008:283040 USPATFULL

TI Component for Machine Structure, Method of Producing the Same and

Material for Induction Hardening IN Havashi, Toru, Okavama, JAPAN

Ohmori, Yasuhiro, Okayama, JAPAN
Kurosawa, Nobutaka, Okayama, JAPAN
Matsuzaki, Akihiro, Chiba, JAPAN

Matsuzaki, Akihiro, Chiba, JAPAN Toyooka, Takaaki, Okayama, JAPAN Yamada, Katsumi, Kanagawa, JAPAN

PA JFE STEEL CORPORATION, Chiyoda-ku, Tokyo, JAPAN (non-U.S. corporation)
PI US 20080247900 A1 20081009

AI US 2005-587731 A1 20050705 (11) WO 2005-JP12397 20050705

20061027 PCT 371 date

PRAI WO 2005-JP10200 20040716

DT Utility FS APPLICATION

LREP FRISHAUF, HOLTZ, GOODMAN & CHICK, PC, 220 Fifth Avenue, 16TH Floor, NEW YORK, NY, 10001-7708, US

CLMN Number of Claims: 38

ECL Exemplary Claim: 1

DRWN 5 Drawing Page(s)

AB There is provided a component for machine structure having a hardened layer through an induction hardening in at least a part thereof, and more improving fatigue strengths as compared with the conventional ones, in which the hardened layer has a hardness Hv of not less than 750 and an average grain size of prior austenite grains is not more than 7 µm over a full thickness of the hardened layer.

SUMM TECHNICAL FIELD

This invention relates to a component for machine structure provided in at least a part thereof with a hardened layer through an induction hardening. As the component for machine structure may be mentioned a drive shaft for an automobile, an input shaft, an output shaft, a crankshaft, inner and outer wheels of a constant velocity joint, a hub, a gear and so on.

RELATED ART

Heretofore, in the components for machine structure such as a drive shaft for an automobile, a constant velocity joint and the like, it is usual to ensure fatigue strengths as a characteristic of the component for machine structure such as torsional fatigue strength, bending fatigue strength, rolling fatigue strength, slip rolling fatigue strength and the like by working a hot rolled steel bar into a given shape through hot forging, cutting, cold forging and the like and then subjecting to induction hardening and tempering.

On the other hand, it is strongly demanded to reduce the weight of the component for the automobile from recent environmental problems. From this viewpoint, it is demanded to more improve the fatigue strengths in the components for the automobile.

As the means for improving the fatigue strengths as mentioned above, there are proposed various methods up to now.

For example, it is considered to increase the hardening depth in the induction hardening for improving the torsional fatigue strength. However, the fatigue strength is saturated at a certain depth even in the increase of the hardening depth.

Also, it is effective to improve the grain boundary strength for the improvement of the torsional fatigue strength. From this viewpoint, there is proposed a technique of refining a particle size of prior austenite by dispersing TiC (see, for example, Patent Document 1).

According to the technique disclosed in the above Patent Document 1, it is attempted to refine the particle size of prior austenite by dispersing a great amount of fine TiC in the heating of the induction hardening, so that it is required to solid-solute TiC prior to the hardening, and hence it is adopted to conduct the heating above 1100° C. at the hot rolling step. For this end, it is required to raise the heating temperature in the hot rolling, so that there is a problem that the productivity is poor.

Also, there is still a problem that the recent demand on the fatigue strengths is not sufficiently satisfied even by the technique disclosed in Patent Document 1.

Furthermore, Patent Document 2 proposes a shaft component for machine structure wherein the torsional fatigue strength is improved by restricting a ratio of a depth of the hardened layer CD to a radius of the induction hardened shaft component R (CD/R) to 0.3-0.7 and further controlling a value A defined with CD/R, austenite particle size γf from a surface to 1 mm after the induction hardening, average Vickers hardness in to (CD/R)=0.1 as induction-hardened and average Vickers hardness in a shaft central portion after the induction hardening to a predetermined range.

However, there is a limit in the improvement of the fatigue properties even in the control of CD/R, so that the recent demand on the torsional fatigue strength could not be yet satisfied.

Patent Document 1: JP-A-2000-154819 (claims, paragraph [0008])

Patent Document 2: JP-A-H08-53714 (claims)

DISCLOSURE OF THE INVENTION

Problems to be Solved by the Invention

It is an object of the invention to provide a component for machine structure capable of more improving the fatigue strength as compared with the conventional ones after the induction hardening and a method of producing the same as well as a material for the induction hardening.

Means for Solving Problems

The inventors have made various studies in order to effectively improve the fatigue strengths of steel materials through the induction hardening. Particularly, the inventors have noticed the torsional fatigue strength as a typical example of the fatigue strengths and made detail examinations thereon, and obtained the following discoveries.

(i) Although the fatigue strength is improved by increasing a transgranular strength of a hardened layer through the induction hardening or a hardness thereof, when the hardness is increased to not less than 750 as a Vickers hardness BV, the fracture migrates from the transgranular fracture to a fracture at a grain boundary of prior austenite, so that the fatigue strength is not improved even if the hardness is increased over the above value.

- (ii) The strength of the prior austenite grain boundary can be improved by refining the grain size of the prior austenite in the hardened layer through the induction hardening, and hence the increase of the fatigue strength can be attained in accordance with the increase of the hardness by making the average grain size of the prior austenite to not more than 7 µm even if the hardness Hv is not less than 750.
- (iii) In order than the hardness Hv of the hardened layer is not less than 750, it is effective to increase contents of one or more of C. Si and P in the starting material, while in order to refine the grain size of the prior austenite in the hardened layer through the induction hardening, it is effective that Mo, B and Ti are included in the starting material and the microstructure prior to the induction hardening is made a fine bainite or martensite introduced with a worked strain through cold working and further that a rapid heating is conducted in the induction hardening and a heating temperature is made low and further a residence time above 800° C. is made short.
- (iv) Although the tempering is usually carried out as another means for increasing the strength after the induction hardening, it is possible to increase the transgranular strength by omitting this means.

The invention is based on the above knowledge.

That is, the thrust and construction of the invention are as follows.

- 1. A component for machine structure characterized by comprising a hardened layer through an induction hardening in at least a part thereof, in which the hardened layer has a hardness Hv of not less than 750 and an average grain size of prior austenite grains is not more than 7 μm over a full thickness of the hardened layer.
- 2. A component for machine structure according to the item 1, which has a chemical composition comprising C: 0.3-1.5 mass %, Si: 0.05-3.0 mass %, Mn: 0.2-2.0 mass %, Al: not more than 0.25 mass %, Ti: 0.005-0.1 mass %, Mo: 0.05-0.6 mass %, Si: not more than 0.105-0.6 mass %, S: not more than 0.10 mass % and P: not more than 0.10 mass %, and the remainder being Fe and inevitable impurities, and satisfying at least one of the following equations (1)-(3):

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C>0.7 mass % (1)
Si>1.1 mass % (2)
P>0.02 mass % (3)
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- 3. A component for machine structure according to item 2, wherein a content of Al in the chemical composition is Al: 0.005-0.25 mass %.
- 4. A component for machine structure according to item 2 or 3, wherein the chemical composition further contains one or more selected from Cr: not more than 2.5 mass %, Cu: not more than 1.0 mass %, Ni: not more than 3.5 mass %, Co: not more than 1.0 mass %, Nb: not more than 0.1 mass %, V: not more than 0.5 mass %, Ta: not more than 0.5 mass %, Hf: not more than 0.5 mass % and Sb: not more than 0.15 mass % and
- 5. A component for machine structure according to any one of items 2 to 4, wherein the chemical composition further contains one or more selected from $\mathbb{W}\colon$ not more than 1.0 mass \$, Ca: not more than 0.005 mass \$, Mg: not more than 0.005 mass \$, Te: not more

than 0.1 mass %, Se: not more than 0.1 mass %, Bi: not more than 0.5 mass %, Pb: not more than 0.5 mass %, Zr: not more than 0.10 mass % and REM: not more than 0.1 mass %.

- 6. A component for machine structure according to item 1, which has a chemical composition comprising C: 0.3-1.5 mass \$, Si : 0.05-3.0 mass \$, Mn: 0.2-2.0 mass \$, Al: not more than 0.25 mass \$, Ti: 0.005-0.1 mass \$, Mo: 0.05-0.6 mass \$, B: 0.0003-0.006 mass \$, S: not more than 0.1 mass \$ and P: not more than 0.10 mass \$ and the remainder being Fe and inevitable impurities in which the hardened layer is not subjected to a tempering.
- 7. A component for machine structure according to item 6, wherein a content of Al in the composition is Al: 0.005-0.25 mass %.
- 8. A component for machine structure according to item 6 or 7, wherein the chemical composition further contains one or more selected from Cr: not more than 2.5 mass \$, Cu: not more than 1.0 mass \$, Ni: not more than 3.5 mass \$, Co: not more than 1.0 mass \$, Nb: not more than 0.1 mass \$, V: not more than 0.5 mass \$, Ta: not more than 0.5 mass \$, Hf: not more than 0.5 mass \$ and Sb: not more than 0.015 mass \$, Ta: not more than 0.05 mass \$ and Sb: not more than 0.015 mass \$
- 9. A component for machine structure according to any one of items 6 to 8, wherein the chemical composition further contains one or more selected from W: not more than 1.0 mass %, Ca: not more than 0.005 mass %, Mg: not more than 0.005 mass %, Te: not more than 0.1 mass %, Se: not more than 0.1 mass %, Si: not more than 0.5 mass %, Zr: not more than 0.1 mass % and REM: not more than 0.1 mass %, Zr: not more than 0.1 mass % and REM: not more tha
- 10. A component for machine structure according to any one of items 2 to 9, wherein Mo-based precipitate is dispersed in an amount of not less than 500 per 1 $\mu m. \, \text{sup.} 3$ and an average particle size of the Mo-based precipitate is not more than 20 mm.
- 11. A method of producing a component for machine structure by subjecting at least a part of a starting material having a chemical composition comprising C: 0.3-1.5 mass \$, Si: 0.05-3.0 mass \$, Mn: 0.2-2.0 mass \$, Al: not more than 0.25 mass \$, Ti: 0.005-0.1 mass \$, Mo: 0.05-0.6 mass \$, B: 0.003-0.006 mass \$, S: not more than 0.1 mass \$ and P: not more than 0.10 mass \$, and the remainder being Fe and inevitable impurities, and satisfying at least one of the following equations (1)-(3):

```
C>0.7 mass % (1)
Si>1.1 mass % (2)
P>0.02 mass % (3)
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to an induction hardening at least once, characterized in that either or both of bainite structure and martensite structure in steel microstructure of the starting material prior to the induction hardening is adjusted to not less than 10 volume %, and a ultimate temperature of the induction hardening is not higher than 1000° C.

12. A method of producing a component for machine structure according to item 11, wherein a content of ${\tt Al}$ in the chemical composition

- 13. A method of producing a component for machine structure according to item 11 or 12, wherein the chemical composition further contains one or more selected from Cr: not more than 2.5 mass %, Cu: not more than 1.0 mass %, Ni: not more than 3.5 mass %, Co: not more than 1.0 mass %, Ni: not more than 0.1 mass %, V: not more than 0.5 mass %, Ta: not more than 0.5 mass %, Hf: not more than 0.5 mass % or than 0.5 mass %, Ta: not more than 0.8 mass %, Ta: not more than 0.9 mass %, Ta:
- 14. A method of producing a component for machine structure according to any one of items 11 to 13, wherein the chemical composition further contains one or more selected from W: not more than 1.0 mass %, Ca: not more than 0.005 mass %, Mg: not more than 0.005 mass %, Te: not more than 0.1 mass %, Se: not more than 0.1 mass %, Se: not more than 0.1 mass %, Te: not more than 0.2 mass %, Te: not
- 15. A material for induction hardening used for the formation of a component for machine structure having in at least a part of its surface a hardened layer with an average grain size of prior austenite of not more than 7 μm through an induction hardening, characterized by having a chemical composition comprising C: 0.3-1.5 mass 8 , Si: 0.05-3.0 mass 8 , Mn: 0.2-2.0 mass 8 , Al: not more than 0.25 mass 8 , Ti: 0.005-0.1 mass 8 , Mo: 0.05-0.6 mass 8 , B: 0.0003-0.006 mass 8 , S: not more than 0.1 mass 8 and P: not more than 0.10 mass 8 , and the remainder being Fe and inevitable impurities, and satisfying at least one of the following equations (1)-(3):

C>0.7 mass % (1) Si>1.1 mass % (2) P>0.02 mass % (3)

and having a steel microstructure in which either or both of bainite structure and martensite structure is not less than 10 volume %.

- 16. A material for induction hardening according to item 15, wherein a content of Al in the composition is Al: 0.005-0.25 mass %.
- 17. A material for induction hardening according to item 15 or 16, wherein the chemical composition further contains one or more selected from Cr: not more than 2.5 mass %, Cu: not more than 1.0 mass %, Ni: not more than 3.5 mass %, Co: not more than 1.0 mass %, Nb: not more than 0.1 mass %, V: not more than 0.5 mass %, Ta: not more than 0.5 mass %, Ta: not more than 0.5 mass % and Sb: not more than 0.15 mass %
- 18. A material for induction hardening according to any one of items 15 to 17, wherein the chemical composition further contains one or more selected from W: not more than 1.0 mass %, Ca: not more than 0.005 mass %, Mg: not more than 0.005 mass %, Te: not more than 0.1 mass %, Se: not more than 0.1 mass %, Bi: not more than 0.5 mass %, Br: not more than 0.5 mass %, Zr: not more than 0.1 mass % Xr: not more than 0.1 mas
- 19. A material for induction hardening according to any one of items 15 to 18, wherein Mo-based precipitate is dispersed in an amount of not less than 500 per 1 µm.sup.3 and an average particle size of

the Mo-based precipitate is not more than 20 nm.

Effect of the Invention

According to the invention, there can be stably obtained components for machine structure having excellent fatigue properties exemplifying torsional fatigue property and rolling fatigue property, and hence there are developed considerable effects on the demands such as weight reduction of automobile parts and the like.

DRWD BRIEF DESCRIPTION OF THE DRAWINGS

FIG. 1 is a graph showing an influence of a heating temperature in an induction hardening on a grain size of prior austenite in a hardened layer with respect to Mo-added steel and Mo-free steel:

FIG. 2 is a transmission electron microphotograph of a fine precipitate (Mo-based precipitate) effective for super-refining γ grains;

FIG. 3 is a graph showing a relation between an average grain size of prior austenite and a torsional fatigue strength with respect to Mo-added steel and Mo-free steel;

FIG. 4 is a graph showing a relation between an average grain size of prior austenite and a torsional fatigue strength with or without tampering:

FIG. 5 is a partial section view of a constant velocity joint; and

FIG. 6 is a section view showing a hardened texture layer in an inner wheel of a constant velocity joint.

DETD BEST MODE FOR CARRYING OUT THE INVENTION

The invention will be explained in detail below.

The components for machine structure according to the invention are a drive shaft for an automobile, an input shaft, a crankshaft, inner and outer wheels of a constant velocity joint, a hub, a gear and so on and have various forms and structures every the component. In any case, they have particularly a hardened layer subjected to a hardening in its portion requiring fatigue strengths or a whole thereof, and the hardened layer is important to have a hardness Hv of not less than 750 and an average grain size of prior austenite of not more than 7 μm over a full thickness of the hardened layer.

The findings arriving at the above knowledge will be described below.

[Grain Size of Prior Austenite in Hardened Layer]

When the average grain size of the prior austenite in the hardened layer through the induction hardening exceeds $7~\mu m$, even if the hardeness Hy of the hardened layer is increased to not less than 750 to improve the transgranular strength as mentioned later, the fatigue fracture is caused at a grain boundary of the prior austenite as a starting point. Therefore, grain size of the prior austenite in the hardened layer is required to be not more than $7~\mu m$, preferably not more than $6~\mu m$, more preferably not more than $5~\mu m$, further preferably not more than $3~\mu m$. Because, the intergranular strength becomes considerably strong

as the grain size is refined. In the conventional technique, even if the transgranular strength is increased, the intergranular strength is not increased and becomes a rate-determining step and hence it is not expected to further increase the strength. On the contrary, the intergranular strength is considerably increased by refining the grain size, so that it is expected to further increase the strength.

Now, the average grain size of the prior austenite in the induction hardened portion is measured as follows.

An outermost surface layer in the hardened layer after the induction hardening has a martensite structure of 100% as an area ratio. The region of 100% martensite structure continues to a certain thickness ranging from the surface of the hardened layer, but the surface area of the martensite structure violently decreases after that. In the invention, a region ranging from the surface of the induction hardened portion to a place that the surface area of the martensite structure decreases to 98% is the hardened layer, and an average depth from the surface is a thickness of the hardened layer.

As to the hardened layer, the average grain size of the prior austenite are measured at positions of 1/5, 1/2 and 4/5 of the whole of the thickness from the surface, and when the average grain size of the prior austenite is 7 μm in any positions, it is defined that the average grain size of the prior austenite grains is not more than 7 μm over the full thickness.

Moreover, the average grain size of the prior austenite grains is measured by corroding a section of the hardened layer with a corroding solution, which is formed by dissolving 50 g of picric acid in 500 g of water and adding 11 g of sodium dodecylbenzene sulfonate, 1 g of ferrous chloride and 1.5 g of oxalic acid, and observing the corroded section at 5 places every each position by means of an optical microscope of from 400 magnification (area of one place: 0.10 mm+0.09 mm), and then analyzing with an image analysis equipment.

Incidentally, when the fatigue such as the rolling fatigue is dependent upon only the microstructure in the vicinity of an extremely surface layer, some effect is obtained even when the thickness of the hardened layer is about 1 mm, while in case of the torsional fatigue strength, it is preferable to make the thickness of the hardened layer to not less than 2 mm. It is more preferably not less than 2.5 mm, further preferably not less than 3 mm.

[Vickers Hardness of Hardened Layer]

When the Vickers hardness Hv of the hardened layer is less than 750, the transgranular strength of the hardened layer is weak, so that even if the prior austenite grains are refined, it can not be expected to improve the fatigue strength so as to match with such a refining. That is, even when the austenite grains are refined to increase the intergranular strength as meniioned above, if the transgranular strength is not increased, the transgranular fracture becomes rate-determined step and it can not be expected to increase the static strength and fatigue strengths. In the invention, therefore, the Vickers hardness Hv of the hardened layer (corresponding to the transgranular strength) is required to be not less than 750. Moreover, the upper limit of the Vickers hardness Hv of the hardened layer is not particularly limited, but it is preferable to be not more than 900 because when it exceeds 900, the amount of the elements added becomes large and hence the machinability of the base metal, the cold forgeability and the

resistance to quenching crack lower.

In the invention, the Vickers hardness is an average of values measured when 98 N (10 kgf) is struck to a position of 1/5 from the surface in the thickness of the hardened laver 5 times.

[Chemical Composition]

There will be described a chemical composition suitable for more increasing the transgranular strength of the hardened layer having the aforementioned particle size of prior austenite and Vickers hardness below.

C: 0.3-1.5 mass %

C is an element affecting a greatest influence on the induction hardenability, and contributes to the improvement of the fatigue strengths by increasing the transgranular strength of the hardened layer and more thickening the induction hardened portion. However, when the amount is less than 0.3 mass %, the hardened layer should be considerably increased for ensuring the required torsional fatigue strength, and hence the occurrence of quenching crack becomes remarkable and it is difficult to obtain a bainite structure as mentioned later. While, when it exceeds 1.5 mass %, it becomes inconvenient to ensure the machinability, cold forgeability and resistance to quenching crack. Therefore, the C amount is preferable to be 0.3-1.5 mass %.

Si: 0.05-3.0 mass %

Si increases the transgranular strength of the hardened layer and contributes to the improvement of the fatigue strengths. Further, it is an element useful for obtaining the bainite structure as mentioned later. From this meaning, it is preferable to be not less than 0.05 mass %. However, when it exceeds 3 mass %, its difficult to ensure the machinability and cold forgeability because ferrite is solid-solution hardened, so that it is preferable to be not more than 3 mass %.

Mn: 0.2-2.0 mass %

Mn is an element for improving the induction hardenability and ensuring the thickness of the hardened layer. However, when the amount is less than 0.2 mass %, the effect is poor. Therefore, the Mn amount is preferable to be not less than 0.2 mass %. While, when it exceeds 2.0 mass %, the residual austenite increases after the hardening, which is easy to bring about the lowering of the hardeness of the surface layer portion. Therefore, it is preferable to be not more than 2.0 mass %. Moreover, when the Mn amount is too large, the machinability tends to become disadvantageous, so that it is more preferably not more than 1.2 mass %, further preferably not more than 1.0 mass %.

Al: 0.005-0.25 mass %

Al is an element effective for the deoxidation of steel. Also, it is an element for suppressing the growth of austenite grains in the heating for the induction hardening to refine the induction hardened portion. When it exceeds 0.25 mass %, the effect is saturated and the increase of the cost is rather caused. Therefore, the Al amount is preferable to be not more than 0.25 mass %. Moreover, the above effect of Al is not developed when the

amount is less than 0.001 mass %, so that the amount is more preferably not less than 0.001 mass %. Further, it is preferable to be not less than 0.005 mass %.

Ti: 0.005-0.1 mass %

Ti has an effect that it bonds to N included as an inevitable impurity to form BN with B mentioned later to thereby prevent the fadeaway of the induction hardenability. Therefore, the amount is preferable to be not less than 0.005 mass %. While, when it exceeds 0.1 mass %, a great amount of TiN is formed and tends to lower the fatigue strengths as a starting point of fatigue fracture, so that the Ti amount is preferable to be 0.005-0.1 mass %. Further preferably, it is 0.01-0.07 mass %. Moreover, in order to effectively develop the hardenability of B by surely precipitating solid-soluted N as TiN, it is preferable that the Ti and N amounts are controlled to be Ti (mass %)/N (mass %) ≥ 3.42 .

Mo: 0.05-0.6 mass %

strength.

Mo has an action that austenite is refined by promoting the formation of bainte structure after the hot working in the heating for the induction hardening to refine the hardened layer. Also, it has an action that the growth of austenite grains in the heating for the induction hardening is suppressed to refine the hardened layer. Particularly, when the heating temperature in the induction hardening is 800-1000° C., preferably 800-950° C., the growth of austenite grains can be suppressed remarkably. Further, it is an element effective for the improvement of the hardenability, so that it is also used for the adjustment of the hardenability. In addition, it has an action of suppressing the formation of a carbide to prevent the lowering of the intergranular

Thus, Mo is a very useful element for obtaining the effects of the invention, but when the amount is not less than 0.05 mass %, it is easy to render the average grain size of prior austenite in the hardened layer into not more than 7 µm, so that it is preferable to be not less than 0.05 mass %. While, when the Mo amount exceeds 0.6 mass %, the hardness of the steel material in the hot working for shaping into a component form considerably increases to lower the workability. Therefore, the Mo amount is preferably 0.05-0.6 mass %, more preferably 0.1-0.6 mass %, further preferably 0.3-0.4 mass %.

Moreover, according to the inventors' examinations, the dragging effect through solid-solution atoms (Solute Drug Effect), pinning effect and the like are considered as a possibility on the refining effect of the prior austenite grains through Mo. Although it is not necessarily clear how to take these effects or the other effect, it is confirmed to develop at least the pinning effect. The details will be described later.

B: 0.0003-0.006 mass %

B is useful for refining the grain size of the prior austenite in the hardened layer when the microstructure prior to the induction hardening includes the bainite structure or martensite structure as mentioned later. Also, the addition of a slight amount improves the induction hardenability and thickens the hardened layer and hence has an effect of improving the fatigue strengths. Furthermore, it is preferentially segregated in the grain boundary to reduce P

concentration segregated in the grain boundary, and hence the intergranular strength is increased to improve the fatigue strengths. However, the effects are poor when the amount is less than 0.0003 mass %. While, when it exceeds 0.006 mass %, the effects are saturated and the increase of the cost is rather caused. Therefore, the B amount is preferably 0.0003-0.006 mass %, more preferably 0.0005-0.004 mass %, further preferably 0.0015-0.003 mass %.

- S: not more than 0.1 mass %
- S is an element forming MnS to improve the machinability of steel. When the amount exceeds 0.1 mass %, it is segregated in the grain boundary to lower the intergranular strength. Therefore, the S amount is preferably not more than 0.1 mass %, further preferably not more than 0.6 mass %.
 - P: not more than 0.10 mass %
- P increases the transgranular strength of the hardened layer and contributes to the improvement of the fatigue strengths. However, when the amount exceeds 0.10 mass %, it is segregated in the grain boundary to lower the intergranular strength. Therefore, the P amount is preferable to be not more than 0.10 mass %.

The remainder other than the above elements may be Fe and inevitable impurities. However, it is particularly preferable to adjust the chemical composition so as to satisfy at least one of the following equations (1)-(3):

C>0.7 mass % (1) Si>1.1 mass % (2) P>0.02 mass % (3)

By satisfying either of the equations (1) to (3) can be made the Vickers hardness Hv of the hardened layer to not less than 750 to increase the transgranular strength, whereby the effect of improving the fatigue strengths accompanied with the refining of the average particle size of prior austenite to not more than 7 μm can be developed considerably.

In the invention, the tempering treatment usually conducted after the induction hardening can be omitted. In this case, the temper softening is not caused, so that even if either of the equations (1) to (3) is not satisfied, Hv of not less than 750 can be satisfied in the above chemical composition range. In case of omitting the tempering, therefore, it is not necessarily required to satisfy at least one of the equations (1) to (3).

The inclusion of one or more selected from the following elements into the above chemical composition is effective to further improve the fatigue strengths.

Cr: not more than 2.5 mass %

Cr is effective for the improvement of the hardenability and is an element useful for ensuring the hardened depth, and may be added. However, when the amount is excessive, the carbide is stabilized to promote the formation of the residual carbide, which lowers the intergranular strength and deteriorates the fatigue strengths. Therefore, the Cr amount is desirable to be reduced as far as possible, but is allowed up to 2.5 mass %. Preferably, it is

not more than 1.5 mass %. Moreover, in order to develop the effect of improving the hardenability, it is preferable to be not less than 0.03 mass %.

Cu: not more than 1.0 mass %

Cu is effective for the improvement of the hardenability and also is solid-soluted in ferrite, in which the fatigue strengths are improved by such a solid solution strengthening. Furthermore, the formation of the carbide is suppressed to control the lowering of the intergranular strength due to the carbide and improve the fatigue strengths. However, when the amount exceeds 1.0 mass %, cracks are caused in the hot working, so that the addition amount is not more than 1.0 mass %. More preferably, it is not more than 0.5 mass %. Moreover, the addition amount of less than 0.03 mass % is small in the effect of improving the hardenability and the effect of suppressing the lowering of the intergranular strength, so that it is desirable to be not less than 0.03 mass %. Preferably, it is 0.1-1.0 mass %.

Ni: not more than 3.5 mass %

Ni is an element improving the hardenability and is used in case of adjusting the hardenability. Also, it is an element suppressing the formation of the carbide to suppress the lowering of the intergranular strength due to the carbide and improve the fatigue strengths. However, Ni is a very expensive element, so that when it is added over 3.5 mass %, the cost of steel material increases, and hence the addition amount is not more than 3.5 mass %. Moreover, the addition amount of less than 0.05 mass % is small in the effect of improving the hardenability and the effect of suppressing the lowering of the intergranular strength, so that it is desirable to be not less than 0.05 mass %. Preferably, it is 0.1-1.0 mass %.

Co: not more than 1.0 mass %

Co is an element suppressing the formation of the carbide to suppress the lowering of the intergranular strength due to the carbide and improve the fatigue strengths. However, Co is a very expensive element, so that when it is added over 1.0 mass %, the cost of steel material increases, and hence the addition amount is not more than 1.0 mass %. Moreover, the addition amount of less than 0.01 mass % is small in the effect of suppressing the lowering of the intergranular strength, so that it is desirable to be not less than 0.01 mass %. Preferably, it is 0.02-0.5 mass %.

Nb: not more than 0.1 mass %

Nb has an effect of improving the hardenability but also bonds to C, N in steel to act as a precipitation strengthening element. Also, it is an element improving the resistance to temper softening. The fatigue strengths are improved by these effects. However, when the amount exceeds 0.1 mass \$ these effects are saturated, so that 0.1 mass \$ is an upper limit. Moreover, the addition amount of less than 0.005 mass \$ is small in the precipitation strengthening action and the effect of improving the resistance to temper softening, so that it is desirable to be not less than 0.005 mass \$. Preferably, it is 0.01-0.05 mass \$.

V bonds to C, N in steel to act as a precipitation strengthening element. Also, it is an element improving the resistance to temper softening. The fatigue strengths are improved by these effects. However, when the amount exceeds 0.5 mass %, the effects are saturated, so that it is not more than 0.5 mass %. Moreover, the addition amount of less than 0.01 mass % is small in the effect of improving the fatigue strengths, so that it is desirable to be not less

than 0.01 mass %. Preferably, it is 0.03-0.3 mass %.

Ta: not more than 0.5 mass %

Ta may be added because there are an effect on the delay in the change of microstructure and an effect of preventing the deterioration of fatigue strengths, particularly rolling fatigue strength. However, when the amount is increased over 0.5 mass %, it does not contribute to the improvement of the strength, so that it is not more than 0.5 mass %. Moreover, in order to develop the action of improving the fatigue strengths, it is preferable to be not less than 0.02 mass %.

Hf: not more than 0.5 mass %

Hf may be added because there are an effect on the delay in the change of microstructure and an effect of preventing the deterioration of fatigue strengths, particularly rolling fatigue strength. However, when the amount is increased over 0.5 mass %, it does not contribute to the improvement of the strength, so that it is not more than 0.5 mass %. Moreover, in order to develop the action of improving the fatigue strengths, it is preferable to be not less than 0.02 mass %.

Sb: not more than 0.015 mass %

Sb may be added because there are an effect on the delay in the change of microstructure and an effect of preventing the deterioration of fatigue strengths, particularly rolling fatigue strength. However, when the amount is increased over 0.015 mass %, the toughness is deteriorated, so that it is not more than 0.015 mass %, preferably not more than 0.010 mass %. Moreover, in order to develop the action of improving the fatigue strengths, it is preferable to be not less than 0.005 mass %.

Further, in order to improve the machinability of steel, it is preferable to include the following elements.

W: not more than 1.0 mass %

W is an element of improving the machinability through embrittlement action. However, when it is added over 1.0 mass %, the effect is saturated and the cost disadvantageously increases, so that it is preferable to be not more than 1.0 mass %. Moreover, W is preferable to be not less than 0.005 mass % for the improvement of the machinability.

Ca: not more than 0.005 mass %

Ca forms a sulfide with MnS, which acts as a chip breaker to improve the machinability, so that it may be added, if necessary. However, when the amount is added over 0.005 mass %, the effect is saturated and the cost increases, so that it is not more than 0.005 mass %. Moreover, when it is less than 0.0001 mass %, the effect of improving the machinability is small, so that it is preferable to be not less than 0.0001 mass %.

Mg: not more than 0.005 mass %

Mg is a deoxidation element but also has an effect of improving the machinability as a stress concentration source, so that it may be added, if necessary. However, when the addition is excessive, the effect is saturated and the cost increases, so that it is not more than 0.005 mass %. Moreover, when it is less than 0.0001 mass %, the effect of improving the machinability is small, so that it is preferable to be not less than 0.0001 mass %.

Ta: not more than 0.1 mass %

Se: not more than 0.1 mass %

Se and Te form MnSe and MnTe through bonding with Mn, respectively, which act as a chip breaker to improve the machinability. However, when the amount exceeds 0.1 mass %, the effect is saturated and the cost increases, so that each amount is not more than 0.1 mass %. Also, in order to improve the machinability, it is preferable that Se amount is not less than 0.003 mass % and Te amount is not less than 0.003 mass %, respectively.

Bi: not more than 0.5 mass %

Bi improves the machinability through the fusion, lubrication and embrittlement actions in the cutting, and may be added for this purpose. However, when it is added over 0.5 mass %, the effect is saturated but also the cost increases, so that it is not more than 0.5 mass %. Moreover, when it is less than 0.01 mass %, the effect of improving the machinability is small, so that it is preferable to be not less than 0.01 mass %.

Pb: not more than 0.5 mass %

Pb improves the machinability through the fusion, lubrication and embrittlement actions in the cutting, and may be added for this purpose. However, when it is added over 0.5 mass %, the effect is saturated but also the cost increases, so that it is not more than 0.5 mass %. Moreover, when it is less than 0.01 mass %, the effect of improving the machinability is small, so that it is preferable to be not less than 0.01 mass %.

Zr: not more than 0.01 mass %

Zr forms a sulfide with MnS, which acts as a chip breaker to improve the machinability. However, when it exceeds 0.01 mass %, the effect is saturated but also the cost increases, so that it is not more than 0.01 mass %. Moreover, when it is less than 0.003 mass %, the effect of improving the machinability is small, so that it is preferable to be not less than 0.003 mass % ity

REM: not more than 0.1 mass %

REM forms a sulfide with MnS, which acts as a chip breaker to improve the machinability. However, when it exceeds 0.1 mass %, the effect is saturated but also the cost increases, so that it is not more than 0.1 mass %. Moreover, in order to improve the machinability, REM is preferable to be not less than 0.0001 mass %.

Although the above is described with respect to the preferable chemical composition range, the chemical composition is limited to the above range and further the steel texture prior to the induction hardening is made to the following texture, whereby there can be

obtained the aforementioned average particle size of the prior austenite of not more than 7 $\mu\text{m}.$

That is, it is preferable that the microstructure of the base metal, i.e. microstructure prior to the hardening (corresponding to the microstructure other than that of the hardened layer after the induction hardening) has a bainite structure and/or a martensite structure and a total of either or both of the bainite structure and the martensite structure is not less than 10 volume %. Because, the bainite structure or the martensite structure is a microstructure finely dispersing the carbide therein as compared with a ferrite-pearlite microstructure, so that an area of an ferrite/carbide interface as a nuclear formation site of austenite in the heating and hardening increases to refine the resulting austenite, which effectively contributes to the refining of the grain size in the hardened layer after the hardening. Further, such a refining of the grain size in the hardened layer after the hardening strengths.

Here, it is more preferable that the total of either or both of the bainite structure and the martensite structure is not less than 20 volume %.

Also, it is preferable that an upper limit of a microstructure fraction in the total of either or both of the bainite structure and the martensite structure is about 90 volume %. Because, when the microstructure fraction exceeds 90 volume %, the effect of refining the prior austenite grains in the hardened layer through the hardening is saturated but also the machinability is rapidly deteriorated.

Moreover, as to the refining of the grain size in the hardened layer after the hardening, the martensite structure has an effect approximately equal to that of the bainite structure. However, the bainite structure is advantageous as compared with the martensite structure because the amount of alloying elements added is less from an industrial viewpoint, and the machinability is good, and it is possible to produce at a low cooling rate.

Further, the volume fraction ratio of bainite and martensite is preferable to be generally bainite martensite=100:0-40:60. As the microstructure prior to the hardening, the martensite structure is preferable for refining the grain size of the prior austenite in martensite of the hardened layer after the induction hardening. However, since martensite is hard, if a great amount of martensite is included in the base metal, the machinability lowers. Therefore, the volume fraction ratio of bainite and martensite is preferable to be baintermarensite=100:0-40:60.

Next, the production method of the component for machine structure according to the invention will be explained.

The component for machine structure according to the invention can be produced by subjecting a steel material having the aforementioned chemical composition to a hot working such as bar steel rolling, hot forging or the like to shape into a component and then subjecting at least a part of the component to an induction hardening under a condition of heating temperature: 800-1000° C. In this case, the term "at least a part" means a portion requiring fatigue strengths.

There is the following method for rendering the average grain size of prior austenite in the induction hardened portion into not more than 7 $\,$ mm.

In the hot working, the total working ratio at 800-1000° C. is not less than 80% and the subsequent temperature region of 700-500° C. is cooled at a rate of not less than 0.2° C./s. Under these conditions, the microstructure prior to the hardening can be made to uniform and fine bainite and/or martensite structure (microstructure fraction: not less than 10 volume %). That is, since bainite and martensite are a microstructure finely dispersing the carbide therein as compared with a ferrite-pearlite microstructure, the area of ferrite/carbide interface as a nucleus formation site of austenite increases in the heating for the induction hardening, which is advantageous for refining the resulting austenite. For this end, it is required that the microstructure fraction in the total of either or both of bainite and martensite is not less than 10 volume %. Further, when the cooling rate in the temperature region of 700-500° C. is less than 0.2° C./s, the microstructure fraction in the total of either or both of bainite and martensite can not be made to not less than 10 volume %. More preferably, the cooling rate is not less than 0.5° C./s. Moreover, the volume ratio of bainite and martensite is preferable to be generally bainite:martensite=100:0-40:60 as previously mentioned.

Furthermore, the bainite and/or martensite structure prior to the induction hardening can be more refined by subjecting to a working of not less than 20% at a temperature region of lower than 800° C. Prior to the induction hardening (hereinafter referred to as a second working step), whereby the further refining of the prior austenite grains after the induction hardening can be attained, so that it is preferable to conduct the second working step. The working at the temperature region of lower than 800° C. may be carried out prior to the cooling at the above cooling rate (temperature region of 700-800° C.) in the hot working step, or a separate cold working may be conducted after the cooling or a warm working may be conducted by re-heating at a temperature of not higher than A.sub.1 transformation point. The working below 800° C. is preferable to be not less than 30%.

As the working method are mentioned, for example, cold forging, cold ironing, forming of rolling, shot peening and the like. By conducting the working below 800° C. is refined the bainite or martensite structure prior to the induction hardening, and hence the average grain size of the prior austenite in the hardened layer obtained after the induction hardening becomes finer to more improve the fatigue strengths.

The prior austenite grains having an average grain size of not more than 7 μm are first obtained by combining the adjustment of the microstructure prior to the hardening by the above working and cooling with the following induction hardening conditions.

At first, when the heating temperature is lower than 800° C., the formation of austenite microstructure is insufficient and the hardened layer can not be obtained. On the other hand, when the heating temperature exceeds 1000° C. or when the temperature rising rate at 600-800° C. is less than 300° C./s, the growth of the austenite grains is promoted and at the same time the scattering of the grains becomes large to lower the fatigue strengths. That is, as to the grain size of the prior austenite in the finally obtained hardened layer, it is important how to prevent the grain growth at the austenite zone in the heating for the hardening. When the microstructure prior to the

hardening is a microstructure having fine bainite or martensite as mentioned above, there are many nucleus formation sites for inverse transformation to austenite, so that when the cooling is started while the produced austenite grains do not grow, the average particle size of the prior austenite in the hardened microstructure can be refined. The growth of the austenite grains proceeds as the temperature becomes higher or as the keeping time in the austenite zone becomes longer, in order to prevent the grain growth and finally obtain the prior austenite grains having an average grain size of not more than 7 µm, the ultimate temperature in the heating is not higher than 1000° C. and the temperature rising rate at 600-800° C c. is not less than 300° C./s.

Moreover, the ultimate temperature in the heating is preferable to be $800-950^\circ$ C., and the temperature rising rate at $600-800^\circ$ C. is preferably not less than 700° C./s, more preferably not less than 1000° C./s.

Also, as the residence time above 800° C. in the induction heating becomes long, the austenite grains grow and the grain size of the prior austenite tends to finally exceed over $7~\mu m$, so that the residence time above 800° C. is preferable to be not more than 5 seconds. More preferably, the heating time is not more than 3 seconds.

Moreover, the above effects develop more remarkably in steels containing Mo within the range defined in the invention. That is, FIG. I shows results examined on a relation between a heating temperature in an induction hardening and a grain size of prior austenite in a hardened layer with respect to Mo-added steel and Mo-free steel.

The results shown in FIG. 1 are obtained as follows.

That is, 150 kg of a steel material having a chemical composition shown in the following a-steel, b-steel, c-steel, d-steel and e-steel is melted in a vacuum melting furnace, hot-forged into 150 mm square to form a dummy billet, which is subjected to a hot working of 80% at 850° C. and cooled at 0.7° C./ s within a temperature range of 700-500° C. to form a bar steel rolled material. Further, a part of the bar steels is subjected to a cold working of 20% after the cooling as a second working step.

(a-steel) C: 0.8 mass %, Si: 0.1 mass %, Mn: 0.78 mass %, P: 0.011 mass %, S: 0.019 mass %, Al: 0.024 mass %, Ti: 0.017 mass %, B: 0.0013 mass %, N: 0.0043 mass %, O: 0.0015 mass %, remainder: Fe and inevitable impurity

(b-steel) C: 0.53 mass %, Si: 0.1 mass %, Mn: 0.74 mass %, P: 0.011 mass %, S: 0.019 mass %, Al: 0.024 mass %, N: 0.0039 mass %, Mo: 0.37 mass %, Ti: 0.018 mass %, B: 0.0013 mass %, remainder: Fe and inevitable impurity

(c-steel) C: 0.9 mass %, Si: 0.1 mass %, Mn: 0.78 mass %, P: 0.011 mass %, S: 0.019 mass %, Al: 0.024 mass %, Mo: 0.37 mass %, Ti: 0.017 mass %, B: 0.0013 mass %, N: 0.0043 mass %,

remainder: Fe and inevitable impurity

(d-steel) C: 0.42 mass %, Si: 1.5 mass %, Mn: 0.78 mass %, P: 0.011 mass %, S: 0.019 mass %, Al: 0.024 mass %, Mo: 0.37 mass %, Ti: 0.017 mass %, B: 0.0013 mass %, N: 0.0043 mass %, remainder: Fe and inevitable impurity

(e-steel) C: 0.42 mass %, Si: 0.2 mass %, Mn: 0.78 mass %, F: 0.05 mass %, S: 0.019 mass %, Al: 0.024 mass %, Mo: 0.37 mass %, Ti: 0.017 mass %, B: 0.0013 mass %, N: 0.0043 mass %, remainder: Fe and inevitable impurity

A test specimen for torsional fatigue is sampled from the resulting bar steel, subjected to an induction hardening at a frequency of 10--200~kHz and a heating temperature of $870\text{--}1050^\circ$ C. and further subjected to a tempering in a heating furnace under conditions of 170° C.+30 minutes to obtain a test material. The induction hardening conditions are adjusted so that the temperature rising rate is not less than 300° C./ s and the residence time above 800° C. is not more than 1 second.

The thus obtained test material is subjected to a torsional fatigue test to measure a stress broken at torsion number of $10. \mathrm{sup}.5$ on the torsional stepped test specimen of 18 mm ϕ . Also, the average grain size of prior austenite in the hardened layer through the induction hardening is measured by the previously mentioned method. Further, the Vickers hardness is measured at a position of 1/5 of the full thickness from the surface of the hardened layer. As the Vickers hardness is adopted an average value measured when 98 N (10 kgf) is struck 5 times.

As shown in FIG. 1, in any of the Mo-added steels and Mo-free steels, the grain size of prior austenite in the hardened layer can be made small by lowering the heating temperature in the induction hardening. Particularly, in case of the Mo-added steels, the refining of the grain size in the hardened layer is remarkably attained by rendering the ultimate temperature in the heating to not higher than 1000° C., preferably not higher than 950° C.

Although the reason on this phenomenon is not clear, the following can be presumed in relation to a carbonitride containing Mo and Ti. That is, it is considered that in case of the Mo-added steel, the above Mo-based fine carbonitride is precipitated to develop a strong pinning force, whereby the austenite grains are more refined as compared with the Mo-free steel. However, even in the induction hardening for a short time, it is considered that if the heating temperature largely exceeds over 1000° C., the fine (Mo, Ti).sub.2(C, N) is dissolved to lessen the pinning effect.

Moreover, it can be seen from FIG. 1 that in case of the Mo-added steel, the grain size of the prior austenite can be more refined by adding the second working step (cold working).

Further, the inventors have guessed that in the steels containing Mo, the reason why the average grain size of prior austenite in the hardened layer through the induction hardening can be more refined to improve the fatique strengths is due to the fact that

the fine Mo-based precipitates are highly dispersed to make the above pinning effect large.

Now, the above a-steel is melted, rolled, subjected to forgings of 80% at 850° C. and 25% at 750° C. and then cooled in air (cooling rate in air: 0.8° C./s). Then, a sample for the observation with a transmission electron microscope is taken out from the starting material prior to the induction hardening to observe a state of fine precipitates. The sample for the observation with the transmission electron microscope is proved by taking out a flat plate sample from the central portion of the starting material and subjecting to an electrolytic polishing with a perchloric acid-methanol based electrolyte for thinning. As the observation zone is too thin, the frequency of dropping out the precipitated particles becomes high, while as the zone becomes too thick, it is difficult to recognize the precipitated particles, so that the thickness of the observation zone is adjusted to a range of 50-100 nm. Moreover, the thickness of the sample is estimated from an electron energy loss spectrum.

In FIG. 2 is shown an example of the actually obtained transmission electron microphotograph. Considering that the thickness of the sample in the view field is about 0.1 μ m, it is revealed that fine precipitates having a diameter of about 5-10 nm are dispersed in a high density of about 3000 particles per 1 μ m. sup. 3.

In the induction hardening, a nucleus of austenite is formed from a grain boundary of bainite or martensite, a packet boundary, a carbide and the like, which grows into grains. The above fine precipitates control the movement of the grain boundary faces os as to push a finger (precipitates) into a balloon (grain boundary face) when the austenite grain boundary face arrives at the precipitates and passes therethrough. Such a control of moving the boundary face is called as a pinning. The pinning force is large as the precipitates becomes small when the amount of the precipitates is constant, or is strong as the amount of the precipitates becomes large when the size of the precipitates is constant.

In the induction heating according to the invention, it is guessed that the pinning is caused by the fine precipitates as shown in FIG. 2 to more promote the refining of the average grain size of prior austenite. Furthermore, it is confirmed that the fine precipitates shown in FIG. 2 are existent even in the material after the induction hardening below 1000° C., from which it is considered that the hardly solubility to the heat treatment at a high temperature for a short time effectively acts to the control of the grain growth of austenite in the induction hardening.

Then, the inventors conducted a model calculation by varying a precipitation volume ratio of Mo in order to estimate the influence of the precipitate dispersed state on the average grain size of prior austenite in the induction heating treatment. That is, assuming that the solid solution of Mo into the other precipitation phase is slight, if a precipitation volume ratio: f and average grain size: do fine Mo-based fine precipitates are determined, there is estimated the number of Mo-based fine precipitates per 1 µm. sup.3 (precipitation density) in the case of uniform dispersion precipitation. If the average grain size of the prior austenite is governed by the pinning of the fine precipitation, such a size is inversely proportional to the precipitation density. So, there are examined the grain size and precipitation density developing the pinning effect considering that the grain size and density of the

precipitates in FIG. 2 attain the average grain size of prior austenite of 2 μ m. As a result, it has been found that the number of the precipitates per 1 μ m.sup.3, which is directly effective for the control of the average grain size of the prior austenite, varies in accordance with the volume ratio of the precipitates, but when the volume ratio is, for example, about 0.2-0.4%, the sufficient pinning effect is developed and the preferable range capable of realizing the refining of the prior austenite grains is as follows.

That is, in order to attain the more refining of the prior austenite grains, it is preferable to ensure not less than 500 fine precipitated particles having a diameter of not more than 20 nm. Further, it is desirable to ensure not less than 1000 fine precipitates particles having a diameter of not more than 15 nm, more preferably not less than 2000 fine precipitated particles having a diameter of not more than 12 nm

Then, as the residue after the precipitates are extracted from the base metal is identified by an X-ray diffractometry, it is guessed to be mainly hop-type (Mo, Ti).sub.2(C, N).
Further, it has been revealed from results of EDX analysis attached with the transmission electron microscope that the atomic ratio of Mo to Ti is about 8:2 and Mo is a main component. Moreover, the precipitates include a composition deviated from a stoichiometric composition of complete (Mo, Ti).sub.2(C, N). In any case, it is considered to be a composite carbonitride including Mo and Ti.

Now, the (Mo, Ti).sub.2(C, N) precipitate

is known to be relatively hard different from the precipitate of Cu or the like and considered to be high in the ability preventing the pass through the grain boundary face. Also, considering that Mo is fairly larger than Ti in the composition ratio and Mo is a hardly dispersing element, it is not thought that (Mo, Ti).sub.2(C, N) becomes rapidly large even it is kept within a temperature range of about 600-700° C. as a (Mo, Ti).sub.2(C, N) precipitation temperature for a short time. Therefore, in order to increase the precipitation amount of (Mo, Ti).sub.2(C , N) to enhance the distribution density, the newly precipitation of (Mo, Ti).sub.2(C, N) can be expected by isothermally keeping the above temperature range within a range of providing a microstructure fraction of bainite and martensite as mentioned later for a short tim while suppressing the coarsening of the previously precipitated (Mo, Ti).sub.2(C, N) at minimum.

In FIG. 3 is shown a relation between a grain size of prior austenite and a torsional fatigue strength in the hardened layer. As seen from this figure, in the Mo-added steel, the grain size becomes small even in a region that the grain size of the prior austenite is not more than 7 µm and also the fatigue properties are improved. On the contrary, in the Mo-free steel, as the grain size becomes not more than 7 µm, even if the grain size is made further smaller than the above value, the fatigue strengths are not improved. This is considered due to the fact that since the hardness of the hardened layer in the Mo-free steel is lower than that of the Mo-added steel, when the grain size of the prior austenite is refined over a certain level, the fatigue fracture is a transgranular fracture and is not affected by the grain size of the prior austenite.

Furthermore, it has been seen that when any content of Si, P is increased in the Mo-added steel c-steel, d-steel, e-steel), the effect of improving the torsional fatigue strength is large at a region that the grain size of the prior austenite is not more than 7 μm . Therefore, as the Vickers hardness of the hardened layer is examined, it has been proved that the effect of improving the fatigue strengths by the refining of the prior austenite grains becomes very large when Hv of a-steel is 700, Hv of b-steel is 740, Hv of c-steel is 902, Hv of d-steel is 775, Hv of e-steel is 760 and the hardness Hv of the hardened layer is

Then, the inventors have discovered that an increment of the fatigue strengths by the refining of the grain size of the prior austenite can be made large by increasing the hardness of the above hardened layer and get an idea that the transgranular strength can be raised by omitting the tempering treatment after the induction hardening instead of the feature that the transgranular strength is raised by increasing any content of C, Si and P, and hence the fatique strengths are raised by the omission of the tempering treatment.

Now, the torsional fatigue strength is examined on the above asteel and b-steel provided that the tempering treatment is omitted in the step of preparing the specimen for the torsional fatigue test. The Vickers hardness Hv in the a-steel and b-steel omitting the tempering is 740 and 780, respectively.

In FIG. 4 is shown a relation between a grain size of prior austenite and torsional fatique property in the hardened layer for the comparison between tempering and no tempering. From FIG. 4, it is seen that the fatique strengths can be improved even by omitting the tempering.

Thus, the invention can positively adopt the method of conducting no tempering. In the usual high-strength steels, there is a case that cracks are caused in the component as the tempering is omitted. For this end, the tempering treatment after the induction hardening is a usual step. Such cracking is usually an intergranular fracture, which results from the lacking of the intergranular strength. In the invention, however, since the intergranular strength is high through the refining of the prior austenite grains, the cracking is hardly caused even in the omission of the tempering treatment. The omission of the tempering treatment is effective in the deterrence of softening due to the tempering and the reduction of the cost based on the tempering.

EXAMPLE 1

not less than 750.

100 kg of a steel shown in Table 1 is melted and heated to 1200° C. and shaped into a sample for torsional fatigue test under hot working conditions and cold working conditions shown in Table 2. The shaped sample is first subjected to an induction hardening at 1050° C. and subsequently to an induction hardening under conditions shown in Table 2. Moreover, the induction hardening at 1050° C. is not carried out in Test No. 10. Also, in the tests other than Test Nos. 29 and 30, the tempering of 160° C.+1 h. is carried out after the induction hardening. The grain size of prior austenite and hardness in the induction hardened portion are measured in the same method as mentioned above. In the torsional fatigue test, stress broken at torsion number of 10.sup.5 is measured on the torsional stepped test specimen of 18 mm4, Also, the microstructure prior to the induction hardening is observed by means of

an optical microscope to identify the microstructure, and at the same time the microstructure fraction (volume %) in one or total of bainite and martensite is measured.

The above measured results are also shown in Table 2. Nos. 7 and 25 are comparative examples wherein all contents of C, Si and P are low, and it is seen that the fatigue strengths in the invention examples are further improved as compared with these comparative examples. Also, when the content of Mo, B or Ti is insufficient as shown in Nos. 26, 27 and 28, the grain size of the prior austenite becomes coarse and the torsional fatigue strength lowers. In Nos. 26, 27 and 28, the microstructure of the base metal is particularly rendered into ferrite-pearlite, so that the grain size of the prior austenite is coarsened and the fatique strengths lower. Furthermore, when the tempering treatment after the induction hardening is not conducted as shown in Nos. 29 and 30, the fatigue strengths are further improved as compared with steels of Nos. 1 and 7. In steel of No. 31, the total working ratio at 800-1000° C. in the hot working is small, so that the grain size of the prior austenite becomes large and the fatigue strengths are low.

TABLE 1

	of Chemic		osition Mn	(mass	%)				
stee	l C P S	Si Al		Mo	Ti				
	B N		others		11				
	D 14		Others						
A	0.75	0.51	0.77	0.01	0.023	0.024	0.45	0.021	0.0023
	0.0040								
В		0.50	0.78	0.01	0.023	0.023	0.44	0.023	0.0021
	0.0042								
С			0.78	0.01	0.02	3 0.024	0.46	0.02	2
_	0.0022 0								
D	0.42	1.11	0.78	0.01	0.023	0.023	0.45	0.023	0.0022
E	0.0044	1.25	0.77	0.01	0.022	0.023	0.45	0.023	0.0023
ы	0.0044	1.23	0.//	0.01	0.022	0.023	0.45	0.023	0.0023
F	0.42	1.49	0.78	0.01	0.023	0.024	0.46	0.022	0.0021
	0.0042								
G	0.48	0.50	0.78	0.02	0.023	0.023	0.44	0.021	0.0021
	0.0044								
H	0.42	0.50	0.78	0.06	0.022	0.024	0.46	0.023	0.0022
_	0.0040								
I	0.41	0.50	0.77	0.10	0.022	0.023	0.45	0.023	0.0022
J	0.0044	0.51	0.77	0.06	0.023	0.024	0.45	0.021	0.0023
J	0.0040	0.51	0.77	0.00	0.023	0.024	0.43	0.021	0.0023
K	0.41	1.25	0.77	0.06	0.022	0.023	0.45	0.023	0.0023
	0.0044								
L	0.70	1.13	0.78	0.02	0.023	0.023	0.44	0.023	0.0021
	0.0042								
M	0.70				0.023	0.023	0.44	0.023	0.0021
	0.0042 C								_
N	0.70						0.44	0.02	3
0	0.0021 0					V: 0.023	0.44	0 022	0.0007
U	0.0042 C		0.78	0.01	0.023	0.023	0.44	0.023	0.0021
P	0.70		0.78	0.01	0.02	3 0.023	0.44	0.02	3
-	0.0021 0				3.02	0.025	3.44	3.02	•
Q	0.70				0.023	0.023	0.44	0.023	0.0021

		Zr: 0.05							
R				0.01	0.023	0.023	0.44	0.023	0.0021
		Pb: 0.02							
S					0.02	3 0.023	0.44	0.02	3
		0.0042							
T				0.01	0.023	0.023	0.44	0.023	0.0021
		Se: 0.01							
U				0.01	0.023	0.023	0.44	0.023	0.0021
		Te: 0.01							
V	0.70			0.01	0.023	0.023	0.44	0.023	0.0021
		Ca: 0.00							
W				0.01	0.023	0.023	0.44	0.023	0.0021
		REM: 0.0							
X						0.023	0.44	0.023	0.0021
		Cu: 0.3,							
Y			0.79	0.011	0.022	0.025	0.45	0.021	0.0024
	0.0041								
Z			0.78	0.01	0.023	0.023	0.03	0.023	0.0021
	0.0042								
AA	0.70	0.50	0.78	0.01	0.023	0.023	0.44	0.023	0.0002
	0.0042								
AB		0.50	0.78	0.01	0.023	0.023	0.44	0.002	0.0021
10	0.0042		0.70	0 01	0.000	0.000	0 11	0.050	
AC		1.12		0.01	0.023	0.002	0.41	0.050	0.0022
		Ca: 0.00		0 01	0.050	0.000	0 11	0.050	0 0000
AD	1.21		0.78	0.01	0.050	0.002	0.41	0.050	0.0022
AE		Ca: 0.00		0.01	0.005	*0.001	0 11	0.050	0 0000
AL				0.01	0.035	<0.001	0.41	0.050	0.0022
	0.0044	Ca: 0.00	3						

TABLE 2

Hot working conditions
Total
Microstructure of nonworking Cold Induction hardening
conditions hardened portion
Total Working Higher

	conditio	ns	hardened	d port	ion		-
		ratio	Cooling	wor	king		Highest
	Residenc	e			Fraction		
		above	rate of		dition	Temperature	Heating
	time				of baini	te +	
Test			500-700° C. I				
			perature al	oove	M.	icrostructure	
	martensi						
No.			(° C./s) rat	io			
		C./s) (°					
	800° C.	(s) const	ruction	(%	5)		
1	A	80	1.0	40		500	890
1	î		rlite +		90	300	030
	_	baini			50		
2	В	80	1.0	40		500	880
_	ī		eutectoid		90	000	000
			de + bainite				
		pearl					
3	C	80	1.0	30		500	880
	1	Pro	eutectoid		90		
		carbi	de + bainite	+			
		pearl	ite				
4	D	80 ~	1.0	20		500	880

	1		bainite +		80		
	-		ferrite +				
			pearlite				
5	E	80	1.0	5		500	885
	1		bainite +		80		
			ferrite +				
			pearlite				
6	F	80	1.0	5		500	890
	1		bainite +		80		
			ferrite +				
7		0.0	pearlite	4.5		500	0.715
/	G 1	80	1.0 bainite +	15	80	500	875
	1		ferrite +		00		
			pearlite				
8	н	80	1.0	15		500	875
	1		bainite +		80	000	0.0
	-		ferrite +				
			pearlite				
9	I	80	1.0	25		500	890
	1		bainite +		80		
			ferrite +				
			pearlite				
10	J	80	1.0	5		500	875
	1		pearlite +		90		
			bainite				
11	K	80	1.0	5		500	865
	1		bainite + ferrite +		80		
12	L	80	pearlite 1.0	25		500	875
12	1	00	pearlite +	2.5	90	300	075
	-		bainite				
13	M	8.0	1.0	5		500	880
	1		pearlite +		90		
			bainite				
14	N	80	1.0	10		500	875
	2		pearlite +		90		
			bainite				
15	0	80	1.0	0		500	875
	1		pearlite +		90		
16	P	80	bainite	40		E00	880
10	1	80	1.0 pearlite +	40	90	500	880
	1		bainite		50		
17	Q	80	1.0	25		500	865
	1	00	pearlite +		90	300	000
			bainite				
18	R	80	1.0	25		500	870
	1		pearlite +		90		
			bainite				
19	S	80	1.0	15		500	880
	1		pearlite +		90		
	_		bainite	0.0			
20	T	80	1.0	35	0.0	500	890
	4		pearlite +		90		
21	U	80	bainite 1.0	40		500	885
41	1	00	pearlite +	40	90	500	003
	-		bainite				
22	V	80	1.0	20		500	865
	1		pearlite +		90		

23	W 1	80	bainite 1.0 pearlite +	15	90	500	875	
0.1		0.0	bainite	0	90	500	0.05	
24	х 1	80	1.0 pearlite + bainite	0	90	500	865	
25	Y 1	80		35	80	500	890	
26	Z	80		20		500	880	
27	1 AA	80	ferrite + pearlite	2.5	0	E00	000	
21	1	80	1.0 ferrite + pearlite	35	0	500	890	
28	AB 1	80	1.0 ferrite +	20	0	500	880	
29	A 1	80	pearlite 1.0 pearlite +	20	90	500	880	
30	G 1	80	bainite 1.0 bainite +	15	80	500	875	
	-		ferrite + pearlite		00			
31	A 1	50	1.0 pearlite + bainite	10	90	500	890	
32	AC 1	80		0	80	300	950	
33	AD 1	80	Proeutectoid carbide +	0	80	300	950	
34	AE	80	bainite + pearlite 0.5	0		300	950	
54	1		Proeutectoid carbide + bainite + pearlite	Ü	80	300	330	
	hardened por Avera	tio	ure of Fatigue n property Torsional Fatigue					
	Vickers	Size	e of Strength				Test	
			Prior (+105)				No.	(Hv)
	γ (μπ)	(MPa)) Remarks					000
	3.4	94	l Example				2	901
	2 4	959	Invention	Evar	mnle		4	201

2.4 959 Invention Example

Invention Example

3.5 989

3 917

751

1.8	890	Invention Example		
1.8	892	Invention Example	5	767
2.6	900	Invention Example	6	772
2.2	800	Comparative Example	7	695
2.1	840	Invention Example	8	750
2.7	869	Invention Example	9	763
3.4	922	Invention Example	10	833
2.8	910	Invention Example	11	783
2.7	976	Invention Example	12	764
2.7	925	Invention Example Invention Example	13	768
		-	14	764
3.2	933	Invention Example	15	764
1.8	930	Invention Example	16	767
3.2	943	Invention Example	17	761
1.6	936	Invention Example	18	759
3.1	935	Invention Example	19	754
1.7	936	Invention Example	20	754
3.8	934	Invention Example	21	756
3.1	920	Invention Example	22	752
2.7	929	Invention Example	23	750
2.3	930	Invention Example	24	770
1.9	935	Invention Example	25	694
2.0	805	Comparative Example	26	690
9.7	701	Comparative Example	27	673
8.7	729	Comparative Example	28	694
8.5	741	Comparative Example	29	820
3.4	990	Invention Example	30	752
2.2	890	Invention Example	31	823
8.1	780	Comparative Example		
5.0	930	Invention Example	32	760
5.3	890	Invention Example	33	900
5.2	930	Invention Example	34	905

As a component for machine structure according to the invention is prepared a constant velocity joint 12 interposed for transmitting a motive energy from a drive shaft 10 to a wheel hub 11 as shown in FIG.

The constant velocity joint 12 is a combination of an outer wheel 13 sand an inner wheel 14. That is, the inner wheel 14 is swingably fixed to an inside of a mouth portion 13a of the outer wheel 13 through balls 15 fitted into a ball track groove formed in the inner face of the mouth portion 13a, while the drive shaft 10 is connected to the inner wheel 14. A stem portion 13b of the outer wheel 13 is, for example, spline-bonded to the hub 11, whereby the motive energy is transmitted from the drive shaft 10 to the hub 11 of the wheel.

A starting material of steel having a chemical composition shown in Table 3 is melted in a converter and continuously cast to form a cast bloom. The cast bloom has a size of 300+400 mm. The cast bloom is rolled into a billet of 150 mm square through a break down step and then rolled into a bar steel of 55 mm .

Then, the bar steel is cut into a given length and hot-forged to form an inner wheel of a constant velocity joint (outer diameter: 45 mm and inner diameter: 20 mm), and thereafter a groove for spline bond is formed on a fitting face thereof by machining or roll forming. Also, a rolling face for balls is formed by machining or cold forging. The cooling after the hot forging is carried out under conditions shown in Table 4. At this moment, the total working ratio in the hot forging and cold forging is obtained by adjusting an area reducing ratio of a section perpendicular to an axial direction of the rolling face.

As shown in FIG. 6, the rolling face 14a for balls lying to the outer wheel of the constant velocity joint is rendered into a hardened texture layer 16 by subjecting to an induction hardening at 1050° C. and to a subsequent induction hardening under conditions shown in Table 4 in an induction hardening apparatus with a frequency of 15 Hz, which is tempered under conditions of 180° C .+2 h in a heating furnace. Moreover, the tempering is omitted in a part of the constant velocity joints. The drive shaft is fitted onto the fitting face of the inner wheel of the thus obtained constant velocity joint, which is mounted to the mouth portion of the outer wheel of the constant velocity joint through balls (steel balls), while the hub is fitted into the stem portion of the outer wheel of the constant velocity joint, whereby there is obtained a constant velocity joint unit (see FIG. 5). Moreover, the specifications of the ball, outer wheel, drive shaft and hub are shown as follows.

Ball: hardened and tempered steel of high carbon

chromium bearing steel SUJ2

Outer Wheel: induction hardened and tempered steel of carbon steel for machine structure Drive Shaft: induction hardened and tempered steel of carbon

Hub: induction hardened and tempered steel of carbon

steel for machine structure

steel for machine structure

Then, the durable test for rolling fatigue strength in the rolling face for balls is carried out in a motive energy transmission system of the constant velocity joint in which a rotational motion of the drive shaft is transmitted to the inner wheel of the constant velocity joint and then through the inner wheel to the hub.

The test for the rolling fatigue strength is carried out by the motive energy transmission under conditions that a torque is 900 N .multidot.m and an operating angle (an angle defined between axial line of inner wheel and axial line of drive shaft) is 20° and a revolution number is 300 rpm, and a time causing the peeling in the rolling face of the inner wheel of the constant velocity joint is evaluated as the rolling fatigue strength. Moreover, the dimensions and forms of the drive shaft, outer wheel of constant velocity joint and the like are set so that the inner wheel of the constant velocity joint becomes weakest in the durable test.

Also, the average grain size of prior austenite and hardness in the hardened layer with respect to the inner wheel of the constant velocity joints prepared under the same conditions are measured by the same methods as mentioned above.

In Table 4 are also shown these results.

TABLE 3

Symbol	Chemica	al comp	osition	(mass	%)				
of s	teel C	Si	Mn	P					
	S A		Mo	Ti	В				
	N	others							
A	0.75 0.0041	0.49	0.75	0.01	0.023	0.023	0.45	0.022	0.0021
В		0.50	0.78	0.01	0.023	0.023	0.44	0.023	0.0022
С		0.50	0.78	0.01	0.02	3 0.024	0.46	0.02	l
	0.0021	0.004	4						
D	0.41	1.10	0.78	0.01	0.023	0.024	0.45	0.023	0.0022
	0.0044								
E		1.30	0.77	0.01	0.022	0.023	0.45	0.022	0.0023
	0.0044								
F		1.50	0.78	0.01	0.023	0.024	0.46	0.022	0.0021
	0.0042								
G		0.49	0.78	0.02	0.023	0.023	0.44	0.023	0.0023
	0.0044	0 50					0.46		
Н	0.41	0.50	0.78	0.06	0.022	0.024	0.46	0.023	0.0022
т	0.0040	0 50	0 22	0.10	0.000	0.022	O 4E	0.000	0 0000
I	0.42	0.50	0.77	0.10	0.022	0.023	0.45	0.023	0.0022
J		0.51	0.77	0.06	0.023	0.024	0.45	0.023	0.0021
J	0.0040	0.31	0.77	0.00	0.023	0.024	0.45	0.023	0.0021
K		1.27	0.77	0.06	0.022	0.024	0.45	0.023	0.0023
10	0.0044	1.27	0.77	0.00	0.022	0.024	0.45	0.023	0.0025
L		1.10	0.78	0.02	0.023	0.023	0.44	0.021	0.0021
_	0.0042	1.10	0	0.02	0.025	0.025	0	0.021	0.0021
M		0.50	0.77	0.01	0.023	0.023	0.44	0.023	0.0021
	0.0040								
N					0.02	3 0.023	0.44	0.02	1
	0.0021	0.004	2 Co:	0.20, 1	Nb:				
	0.0	43, V:	0.18						
0	0.70	0.50	0.77	0.01	0.023	0.023	0.44	0.023	0.0022
	0.0042	Cr: 0.							
P					0.02	3 0.024	0.44	0.02	3
			2 V: (
Q	0.70		0.78	0.01	0.023	0.023	0.44	0.023	0.0021
	0.0042								
R	0.70	0.50	0.79	0.01	0.023	0.023	0.44	0.023	0.0021

	0.0042	Pb: 0.02						
S		0.50 0.78		0.02	3 0.023	0.44	0.02	3
Т	0.0022	0.0042 Bi: 0.50 0.77		0.023	0.024	0.44	0.023	0.0021
	0.0042	Se: 0.01						
U	0.70	0.50 0.78 Te: 0.01	0.01	0.023	0.023	0.44	0.023	0.0021
V	0.70	0.50 0.78	0.01	0.023	0.023	0.44	0.023	0.0021
W	0.0042	Ca: 0.003 0.50 0.79	0.01	0.023	0.024	0.44	0.023	0.0021
W	0.0042	REM: 0.002	0.01	0.023	0.024	0.44	0.023	0.0021
Х	0.70	0.50 0.78		0.023	0.023	0.44	0.023	0.0021
	0.0042 T	Cu: 0.3, Ni: e: 0.01	0.44,					
Y	0.48	0.51 0.79	0.01	0.022	0.024	0.45	0.021	0.0024
_	0.0041	0.50 0.78	0.01	0.023	0.023	0.02	0.022	0.0001
Z	0.70	0.50 0.78	0.01	0.023	0.023	0.02	0.022	0.0021
AA		0.50 0.78	0.01	0.023	0.025	0.44	0.023	0.0002
AB	0.0042	0.50 0.78	0.01	0.023	0.023	0.44	0.002	0.0021
	0.0042							
AC		1.12 0.78	0.01	0.023	0.002	0.41	0.050	0.0022
AD	0.0044 1.21	Ca: 0.003 0.60 0.78	0.01	0.050	0.002	0.41	0.050	0.0022
	0.0044	Ca: 0.003						
AE		0.60 0.78 Ca: 0.003	0.01	0.035	<0.001	0.41	0.050	0.0022
	0.0011	Ca. 0.005						
TABLE	4							
		Hot work	ing					
		ditions						
	Microstr	ucture of Total						
	Microstr	ucture of non-		h	ardened p			
	conditio	working	hardene	d norti		Ind	uction	hardening
	Average		Fatigu		OII			
			Coolin		working			Highest
	Residenc property			F.	raction			Grain
			rate o		conditio		peratur	
	Time	on Dolling		0	f bainite	+		size of
	Symb	or Rolling ol 800°C.	500-700	° C.				
	Working	rising rat	e Temp	peratur	e Above			structure
Test N	martensi				γ absence	of	Fatigue	
1000 14	ratio (%) (° C./s) (° (c.)					
						n ,		
	nardness	(Hv) (μm)	temperi	ng	life (hr)	Kemark	5	
1	A		1.0		40	500		890
	1 presence	Pearlite			0	823		3.4
	presence	163 baini		CIOU EX	ашъте			
2	В	80	1.0		40	500		880
	1 presence	Proeutec 177			0 ample	901		2.4
	bresence	carbi						

carbide + bainite +

		pearlite		
3	C	80 1.0 30	500	
	880	1 proeutectoid	90	917
	3.5	presence 173 Invention	Example	
		carbide + bainite + pearlite 80 1.0 20 bainite + 80 151 Invention Example		
4		pearlite	F 0.0	885
4	D 1	80 1.0 20	500 751	1.8
	presence	151 Invention Example	731	1.0
	properies	ferrite +		
		pearlite		
5	E	80 1.0 5	500	890
	1	ferrite + pearlite 80 1.0 5 bainite + 80 148 Invention Example	767	1.8
	presence	148 Invention Example		
		ferrite + pearlite		
6	F	pearlite	500	875
0	1	80 1.0 5 bainite + 80 155 Invention Example	772	2.6
	presence	155 Invention Example	7.72	2.0
	P# 00 01100	ferrite + pearlite 80 1.0 15 bainite + 80 141 Comparative		
		pearlite		
7	G	80 1.0 15	500	875
	1	bainite + 80	695	2.2
	presence	141 Comparative		
		rerrite +		
	Example	noarlito		
8	н	pearlite 80 1.0 15 bainite + 80	500	890
	H 1	bainite + 80	750	2.1
	presence			
		ferrite +		
		ferrite + pearlite 80 1.0 25 bainite + 80		
9	1	80 1.0 25	500	875
	presence	152 Invention Example	763	2.7
	presence	ferrite +		
		pearlite		
10	J 1	ferrite + pearlite 80 1.0 5 pearlite + 90	500	865
	1	pearlite + 90	833	3.4
	presence			
		bainite 80 1.0 5 bainite + 80 150 Invention Example		
11	K 1	80 1.0 5	500	875
	presence	Dainite + 80	783	2.8
	bresence	ferrite +		
		ferrite + pearlite 80 1.0 25 Pearlite + 90		
12	L	80 1.0 25	500	880
	±	Pearlite + 90	764	2.7
	presence	181 Invention Example bainite 80 1.0 5 Pearlite + 90 165 Invention Example bainite		
		bainite	500	0.775
13	M	80 1.0 5	500 768	875 2.7
	nrecence	165 Invention Example	768	2.1
	bresence	bainite		
14	N	165 Invention Example bainite 80 1.0 10 2 pearlite + presence 164 Invention bainite 80 1.0 0 pearlite + 90 166 Invention Example bainite	500	
	875	<pre>2 pearlite +</pre>	90	764
	3.2	presence 164 Invention	Example	
		bainite		
15	0	80 1.0 0	500	880
	1	pearlite + 90	764	1.8
	presence	166 Invention Example bainite		
		Dainite		

16	D.	80 1 presence bainit 80 pearlite 168 bainit 80 pearlite 161 bainit	1 0	40	E00	
10	865	1	1.0 n	earlite +	90	767
	3.2	presence	172	Invention	Example	707
		bainit	e		•	
17	Q	80	1.0	25	500	865
	1	pearlite	+_	90	761	1.6
	presence	168	Inven	tion Example		
18	D	on Dailli	1 n	25	500	870
10	1	pearlite	+	90	759	3.1
	presence	161	Inven	tion Example		
	-	bainit	:e	_		
19	S	80	1.0	15	500	
	880	1	150 P	earlite +	90	754
	1./	bainit 80 1 presence bainit 80 pearlite 160	159	Invention	Example	
20	т	80	1 0	35	500	890
	4	pearlite	+	90	754	3.8
	presence	160	Inven	tion Example		
		bainit	e	40 90		
21	U 1	80	1.0	40	500	885
	1	pearlite	+	90 tion Example	756	3.1
	presence	103	Inven	tion Example		
22	V	80	1.0	20 90	500	865
	1	pearlite	+	90	752	2.7
	presence	164	Inven	tion Example		
23	W	80	1.0	15 90	500	875
	1	Pearlite	+	90 tion Example	750	2.3
	presence		- 6	=		
24	Х	80	1.0	90	500	865
	1	pearlite	+	90	770	1.9
	presence	160	Inven	tion Example		
		bainit	e			
25	Y	80	1.0	35 80	500	890
	1 presence	pearlite	Compa	rative	694	2.0
	presence	bainit	.e comba	Iacive		
	Example					
26	Z	80	1.0	20	500	880
	1	ferrite -	+	0	690	9.7
	presence	120	Compa	rative		
	Example	pearl:				
27	AA	80	1 0	35 0 rative	500	890
- '	1	Ferrite -	+	0	673	8.7
	presence	131	Compa	rative		
		pearl:	ite			
	Example					
28	AB 1	80	1.0	20	500	880 8.5
	presence	rerrite -	Compa	20 0 rative	694	8.5
	presence	pearl:	ite	Ideave		
	Example					
29	Example A	80	1.0	20 90	500	880
			+_	90	820	3.4
	absence	180	Inven	tion Example		
30	G	painit 80	1 0	15	500	875
50	1	bainit 80 Bainite	ferri	te + 80	752	2.2
	-					

	absence	150	Invent	ion Example		
		pea	arlite			
31	A	50	1.0	10	500	890
	1	pearl:	ite +	90	823	8.1
	presence	140	Compar	ative		
		ba:	inite			
	Example					
32	AC	70	0.7	0	300	950
	1	bainit	te + ferrit	e + 80	760	5.0
	presence	162	Invent	ion Example		
		pea	arlite			
33	AD	70	0.7	0	300	950
	1	Proeut	tectoid	80	900	5.2
	presence	179	Invent	ion Example		
		car	rbide + bai	nite +		
		pea	arlite			
34	AE	70	0.7	0	300	950
	1	Proeut	tectoid	80	906	5.2
	presence	173	Invent	ion Example		
		car	rbide + bai	nite +		
		pea	arlite			

As seen from Table 4, Nos. 7 and 25 are comparative examples wherein all contents of C. Si and P are low, and the rolling fatique life is improved in the invention examples as compared with these comparative examples. Also, when the content of Mo, B or Ti is lacking as in Nos. 26, 27 and 28, the grain size of the prior austenite is coarsened and the rolling fatigue life lowers. Further, in Nos. 26, 27 and 28, the microstructure of the base material is particularly ferrite-pearlite, so that the grain size of the prior austenite is coarsened and the rolling fatigue life lowers. In addition, when the tempering treatment is omitted after the induction hardening as in Nos. 29 and 30, the rolling fatigue life is further improved as compared with No. 1 steel and No. 7 steel, respectively. Moreover, in No. 31 steel, the total working ration of 800-1000° C. in the hot working is small, so that the grain size of the prior austenite becomes large and the rolling fatique life is low.

CLM What is claimed is:

- 1. A component for machine structure characterized by comprising a hardened layer through an induction hardening in at least a part thereof, in which the hardened layer has a hardness Hv of not less than 750 and an average grain size of prior austenite grains is not more than 7 μm over a full thickness of the hardened layer.
- 2. A component for machine structure according to claim 1, which has a chemical composition comprising C: 0.3-1.5 mass \$, Si: 0.05-3.0 mass \$, Mn: 0.2-2.0 mass \$, Al: not more than 0.25 mass \$, Ti: 0.005-0.1 mass \$, Mo:10 0.05-0.6 mass \$, 8: 0.0003-0.06 mass \$, S: not more than 0.1 mass \$ and P: not more than 0.10 mass \$, and the remainder being Fe and inevitable impurities, and satisfying at least one of the following equations (1)-(3): C>0.7 mass \$ (1) Si>1.1 mass \$ (2) P>0.02 mass \$ (3)
- 3. A component for machine structure according to claim 2, wherein a content of Al in the chemical composition is Al: 0.005-0.25 mass %.
- $4\,.$ A component for machine structure according to claim 2, wherein the chemical composition further contains one or more selected from

Cr: not more than 2.5 mass %, Cu: not more than 1.0 mass %, Ni: not more than 3.5 mass %, Co: not more than 1.0 mass %, Nb: not more than 0.1 mass %, V: not more than 0.5 mass %, Ta: not more than 0.5 mass %, Hf: not more than 0.5 mass % and Sb: not more than 0.015 mass %.

5. A component for machine structure according to claim 2, wherein the chemical composition further contains one or more selected from W: not more than 1.0 mass %, Ca: not more than 0.005 mass %, Mg: not more than 0.005 mass %, Te: not more than 0.1 mass %, Se: not more than 0.1 mass %, Bi: not more than 0.5

mass %, Pb: not more than 0.5 mass %, Zr: not more than 0.01 mass % and REM: not more than 0.1 mass %.

- 6. A component for machine structure according to claim 1, which has a chemical composition comprising C: 0.3-1.5 mass %, Si : 0.05-3.0 mass %, Mn: 0.2-2.0 mass %, Al: not more than 0.25 mass %, Ti: 0.005-0.1 mass %, Mo: 0.05-0.6 mass %, B: 0.0003-0.006 mass %, S: not more than 0.1 mass % and P: not more than 0.10 mass % and the remainder being Fe and inevitable impurities in which the hardened laver is not subjected to a tempering.
- 7. A component for machine structure according to claim 6, wherein a content of Al in the composition is Al: 0.005-0.25 mass %.
- 8. A component for machine structure according to claim 6, wherein the chemical composition further contains one or more selected from Cr: not more than 2.5 mass %, Cu: not more than 1.0 mass %, Ni: not more than 3.5 mass %, Co: not more than 1.0 mass %, Nb: not more than 0.1 mass %, V: not more than 0.5 mass %, Ta: not more than 0.5 mass %, Hf: not more than 0.5 mass % and Sb: not more than 0.015 mass %.
- 9. A component for machine structure according to claim 6, wherein the chemical composition further contains one or more selected from W: not more than 1.0 mass %, Ca: not more than 0.005 mass %, Mg: not more than 0.005 mass %, Te: not more than 0.1 mass %, Se: not more than 0.1 mass %, Bi: not more than 0.5 mass %, Pb: not more than 0.5 mass %, Zr: not more than 0.01 mass % and REM: not more than 0.1 mass %.
- 10. A component for machine structure according to claim 2, wherein Mo-based precipitate is dispersed in an amount of not less than 500 per 1 µm3 and an average particle size of the Mo-based precipitate is not more than 20 nm.
- 11. A method of producing a component for machine structure by subjecting at least a part of a starting material having a chemical composition comprising C: 0.3-1.5 mass %, Si: 0.05-3.0. mass %, Mn: 0.2-2.0 mass %, Al: not more than 0.25 mass %, Ti: 0.005-0.1 mass %, Mo: 0.05-0.6 mass %, B: 0.0003-0.006 mass %, S: not more than 0.1 mass % and P: not more than 0.10 mass %, and the remainder being Fe and inevitable impurities, and satisfying at least one of the following equations (1)-(3): C>0.7 mass % (1) Si>1.1 mass % (2) P>0.02 mass % (3) to an induction hardening at least once, wherein either or both of bainite structure and martensite structure in steel microstructure of the starting material prior to the induction hardening is adjusted to not less than 10 volume %, and an ultimate temperature of the induction hardening is not higher

than 1000° C.

- 12. A method of producing a component for machine structure according to claim 11, wherein a content of Al in the chemical composition is Al: 0.005-0.25 mass %.
- 13. A method of producing a component for machine structure according to claim 11, wherein the chemical composition further contains one or more selected from Cr: not more than 2.5 mass %, Cu: not more than 1.0 mass %, Ni: not more than 3.5 mass %, Co: not

more than 1.0 mass %, Nb: not more than 0.1 mass %, V: not more than 0.5 mass %. Ta: not more than 0.5 mass %. Hf: not

- more than 0.5 mass %, Ta: not more than 0.5 mass %, Hf: not more than 0.5 mass % and Sb: not more than 0.015 mass %.
- 14. A method of producing a component for machine structure according to claim 11, wherein the chemical composition further contains one or more selected from W: not more than 1.0 mass %, Ca: not more than 0.005 mass %, Mg: not more than 0.005 mass %, Te: not more than 0.1 mass %, Se: not more than 0.1 mass %, Bi: not more than 0.5 mass %, Te: not more than 0.5 mass %, Te: not more than 0.1 mass % and REM: not more than 0.1 mass % as Well mass % and REM: not more than 0.1 mass % as Well mass % and REM: not more than 0.1 mass % as Well mass % and REM: not more than 0.1 mass % as Well mass % and REM: not more than 0.1 mass % as Well mass % and REM: not more than 0.1 mass % as Well mass % and REM: not more than 0.1 mass % as Well mass % and REM: not more than 0.1 mass % as Well mass % and REM: not more than 0.1 mass % as Well mass % and REM: not more than 0.1 mass % as Well mass % and REM: not more than 0.1 mass % as Well mass % and Mell mass % an
- maicrostructure in which either or both of bainite structure and martensite structure is not less than 10 volume %.
- 16. A material for induction hardening according to claim 15, wherein a content of Al in the composition is Al: 0.005-0.25 mass \$.
- 17. A material for induction hardening according to claim 15, wherein the chemical composition further contains one or more selected from Cr: not more than 2.5 mass %, Cu: not more than 1.0 mass %, Ni: not more than 3.5 mass %, Co: not more than 1.0 mass %, No: not more than 0.1 mass %, V: not more than 0.5 mass %, Ta: not more than 0.5 mass %, Hf: not more than 0.5 mass % and Sb: not more than 0.015 mass %.
- 18. A material for induction hardening according to claim 15, wherein the chemical composition further contains one or more selected from W: not more than 1.0 mass %, Ca: not more than 0.005 mass %, Me: not more than 0.10 mass %, Te: not more than 0.1 mass %, Se: not more than 0.1 mass %, Bi: not more than 0.1 mass % and REM: not more than 0.1 mass % and REM: not more than 0.1 mass % and
- 19. A material for induction hardening according to claim 15, wherein Mo-based precipitate is dispersed in an amount of not less than 500 per 1 μ m.sup.3 and an average particle size of the Mo-based precipitate is not more than 20 nm.

20. A component for machine structure according to claim 3, wherein the chemical composition further contains one or more selected from

Cr: not more than 2.5 mass %, Cu: not more than 1.0

mass %, Ni: not more than 3.5 mass %, Co: not more than 1.0

mass %, Nb: not more than 0.1 mass %, V: not more than 0.5

mass %, Ta: not more than 0.5 mass %, Hf: not more than 0.5 mass % and Sb: not more than 0.015 mass %.

21. A component for machine structure according to claim 20, wherein the chemical composition further contains one or more selected from W: not more than 1.0 mass %, Ca: not more than 0.005 mass %, Mg; not more than 0.005 mass %, Te: not more than 0.1 mass %,

Se: not more than 0.1 mass %, Bi: not more than 0.5

mass %, Pb: not more than 0.5 mass %, Zr: not more than 0.01 mass % and REM: not more than 0.1 mass %.

22. A component for machine structure according to claim 3, wherein the chemical composition further contains one or more selected from W: not more than 1.0 mass %, Ca: not more than 0.005 mass %, Mg: not more than 0.005 mass %, Te: not more than 0.1 mass %, Se: not more than 0.1 mass %, Bi: not more than 0.5

mass %, Pb: not more than 0.5 mass %, Zr: not more than 0.01 mass % and REM: not more than 0.1 mass %.

23. A component for machine structure according to claim 4, wherein the chemical composition further contains one or more selected from W: not more than 1.0 mass %, Ca: not more than 0.005 mass %, Mg: not more than 0.005 mass %, Te: not more than 0.1 mass %, Se: not more than 0.1 mass %, Bi: not more than 0.5

mass %, Pb: not more than 0.5 mass %, Zr: not more than 0.01 mass % and REM: not more than 0.1 mass %.

24. A component for machine structure according to claim 7, wherein the chemical composition further contains one or more selected from Cr: not more than 2.5 mass %, Cu: not more than 1.0

mass %, Ni: not more than 3.5 mass %, Co: not more than 1.0

mass %, Nb: not more than 0.1 mass %, V: not more than 0.5

mass %, Ta: not more than 0.5 mass %, Hf: not more than 0.5 mass % and Sb: not more than 0.015 mass %.

25. A component for machine structure according to claim 24, wherein the chemical composition further contains one or more selected from W: not more than 1.0 mass %, Ca: not more than 0.005 mass %, Mg: not more than 0.005 mass %, Te: not more than 0.1 mass %, Se: not more than 0.1 mass %, Bi: not more than 0.5

mass %, Pb: not more than 0.5 mass %, Zr: not more than 0.01 mass % and REM: not more than 0.1 mass %.

26. A component for machine structure according to claim 7, wherein the chemical composition further contains one or more selected from W: not more than 1.0 mass %, Ca: not more than 0.005 mass %, Mg: not more than 0.005 mass %, Te: not more than 0.1 mass %, Se: not more than 0.1 mass %, Bi: not more than 0.5

mass %, Pb: not more than 0.5 mass %, Zr: not more than 0.01 mass % and REM: not more than 0.1 mass %.

27. A component for machine structure according to claim 8, wherein the chemical composition further contains one or more selected from W: not more than 1.0 mass %, Ca: not more than 0.005 mass %, Mg: not more than 0.005 mass %, Te: not more than 0.1 mass %, Se: not more than 0.1 mass %, Bi: not more than 0.5

mass %, Pb: not more than 0.5 mass %, Zr: not more than 0.01 mass % and

REM: not more than 0.1 mass %.

0.5 mass % and Sb: not more than 0.015 mass %.

- 28. A method of producing a component for machine structure according to claim 12, wherein the chemical composition further contains one or more selected from Cr: not more than 2.5 mass %, Cu: not more than 1.0 mass %, Ni: not more than 3.5 mass %, Co: not more than 1.0 mass %, Ni: not more than 0.1 mass %, V: not more than 0.5 mass %, Ta: not more than 0.5 mass %, Hf: not more than
- 29. A method of producing a component for machine structure according to claim 28, wherein the chemical composition further contains one or more selected from W: not more than 1.0 mass %, Ca: not more than 0.005 mass %, My: not more than 0.005 mass %, Te: not more than no.1 mass %, Se: not more than 0.1 mass %, Te: not more than 0.5 mass %, Te: not more than 0.5 mass %, Te: not more than 0.1 mass %, Te: not more than 0.1 mass %.
- 30. A method of producing a component for machine structure according to claim 12, wherein the chemical composition further contains one or more selected from W: not more than 1.0 mass %, Ca: not more than 0.005 mass %, Mg: not more than 0.005 mass %, Te: not more than 0.1 mass %, Se: not more than 0.1 mass %, Bi: not more than 0.5 mass %, Zr: not more than 0.5 mass %, Zr: not more than 0.1 mass %.
- 31. A method of producing a component for machine structure according to claim 13, wherein the chemical composition further contains one or more selected from W: not more than 1.0 mass %, Ca: not more than 0.005 mass %, Mg: not more than 0.005 mass %, Te: not more than no.1 mass %, Se: not more than no.1 mass %, Bi: not more than 0.5 mass %, Pb: not more than 0.5 mass %, Zr: not more than 0.1 mass % and REM: not more than 0.1 mass % as %.
- 32. A material for induction hardening according to claim 16, wherein the chemical composition further contains one or more selected from Cr: not more than 2.5 mass %, Cu: not more than 1.0 mass %, Ni: not more than 3.5 mass %, Co: not more than 1.0 mass %, No: not more than 0.1 mass %, V: not more than 0.5 mass %, Ta: not more than 0.5 mass %, Hf: not more than 0.5 mass % and Sb: not more than 0.015 mass %.
- 33. A material for induction hardening according to claim 32, wherein the chemical composition further contains one or more selected from W: not more than 1.0 mass %, Ca: not more than 0.005 mass %, Te: not more than 0.1 mass %, Se: not more than 0.1 mass %, Be: not more than 0.1 mass %, Be: not more than 0.1 mass % and REW: not more than 0.1 mass % and REW: not more than 0.1 mass % and
- 34. A material for induction hardening according to claim 16, wherein the chemical composition further contains one or more selected from \mathbb{W} : not more than 1.0 mass \$, Ca: not more than 0.005 mass \$, Mg: not more than 0.005 mass \$, Te: not more than 0.1 mass \$, Se: not more than 0.1 mass \$, Bi: not more than 0.5 mass \$, Pb: not more than 0.01 mass \$ and REM: not more than 0.1 mass \$ and REM: not more than 0.1 mass \$
- 35. A material for induction hardening according to claim 17, wherein the chemical composition further contains one or more selected from W: not more than 1.0 mass \$, Ca: not more than 0.005 mass \$, Mg: not more than 0.005 mass \$, Te: not more than 0.1 mass \$,

Se: not more than 0.1 mass %, Bi: not more than 0.5 mass %, Pb: not more than 0.5 mass %, Zr: not more than 0.01 mass % and REM: not more than 0.1 mass %.

- 36. A material for induction hardening according to claim 16, wherein Mo-based precipitate is dispersed in an amount of not less than 500 per 1 μ m3 and an average particle size of the Mo-based precipitate is not more than 20 nm.
- 37. A material for induction hardening according to claim 17, wherein Mo-based precipitate is dispersed in an amount of not less than 500 per 1 µm3 and an average particle size of the Mo-based precipitate is not more than 20 nm.
- 38. A material for induction hardening according to claim 18, wherein Mo-based precipitate is dispersed in an amount of not less than 500 per 1 μ m3 and an average particle size of the Mo-based precipitate is not more than 20 nm.

INCL INCLM: 420/084.000

INCLS: 148/567.000; 420/121.000; 420/089.000

NCL NCLM: 420/084.000

NCLS: 148/567.000; 420/089.000; 420/121.000

IC IPCI C22C0038-00 [I,A]; C21D0001-42 [I,A]; C22C0038-20 [I,A]
CAS INDEXING IS AVAILABLE FOR THIS PATENT.

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L22 ANSWER 1 OF 30 USPATFULL on STN

PI US 20080247900 A1 20081009

L22 ANSWER 2 OF 30 USPATFULL on STN PI US 20070193658 A1 20070823

L22 ANSWER 3 OF 30 USPATFULL on STN PI US 20070163687 A1 20070719

L22 ANSWER 4 OF 30 USPATFULL on STN PI US 7195736 B1 20070327

L22 ANSWER 5 OF 30 USPATFULL on STN

WO 2001059170

US 20060157163

PI US 20070062619 A1 20070322

L22 ANSWER 6 OF 30 USPATFULL on STN PI US 20060285992 A1 20061221

L22 ANSWER 7 OF 30 USPATFULL on STN PI US 20060239853 A1 20061026

L22 ANSWER 8 OF 30 USPATFULL on STN

L22 ANSWER 9 OF 30 USPATFULL on STN

L22 ANSWER 9 OF 30 USPATFULL on STN PI US 20050265886 A1 20051201

L22 ANSWER 10 OF 30 USPATFULL on STN PI US 20050252580 A1 20051117

L22 ANSWER 11 OF 30 USPATFULL on STN

PI US 20050207687 A1 20050922

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For a list of commands available to you in the current file, enter
"HELP COMMANDS" at an arrow prompt (=>).
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PROCESSING COMPLETED FOR L23
L24
          69 DUP REM L23 (0 DUPLICATES REMOVED)
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PI JP 2009001887 A 20090108 JP 2007-165962 20070625
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PI	JP 2009001864	A 200	90108	JP 2007-163851	20070621
L24	ANSWER 3 OF 69 PATENT NO.	CAPLUS COPYRI KIND DAT	IGHT 2009 FE	ACS on STN APPLICATION NO. 	DATE
PI	JP 2008308727	A 200	81225	JP 2007-157349	20070614
L24	ANSWER 4 OF 69 PATENT NO.	CAPLUS COPYRI	IGHT 2009 TE	ACS on STN APPLICATION NO. JP 2007-154899	DATE
PI	JP 2008308704	A 200	81225	JP 2007-154899	20070612
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PI	JP 2008149340	A 200	80703	JP 2006-338791	20061215
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PI	JP 2008013788	A 200	080124	JP 2006-183512	20060703
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PI	JP 2007031787	A 200	70208	JP 2005-217453	20050727
	ANSWER 10 OF 69 US 2007019365				
	ANSWER 11 OF 69 US 2007016368				

	ANSWER 12 OF 69 US 20070062619	USPATFULL on STN Al 20070322	
L24 PI	ANSWER 13 OF 69 US 7195736 WO 2001059170	USPATFULL on STN B1 20070327 20010816	
	PATENT NO.	CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO.	DATE
PI	JP 2006348321	A 20061228 JP 2005-173112	20050614
L24	ANSWER 15 OF 69 PATENT NO.	CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO.	DATE
PI	JP 2006299296		20050415
L24	ANSWER 16 OF 69 PATENT NO.	CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO. A 20060921 JP 2005-64316	DATE
PI	JP 2006249457	A 20060921 JP 2005-64316	20050308
L24	ANSWER 17 OF 69 PATENT NO.	CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO. A 20060810 JP 2005-21019	DATE
PI	JP 2006206967	A 20060810 JP 2005-21019	20050128
L24	ANSWER 18 OF 69 PATENT NO.	CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO.	DATE
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L24 PI	ANSWER 20 OF 69 US 20060239853	USPATFULL on STN Al 20061026	
		USPATFULL on STN Al 20060720	
L24	ANSWER 22 OF 69 PATENT NO.	CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO	DATE
PI	JP 2005220423	A 20050818 JP 2004-30993	20040206
L24	ANSWER 23 OF 69 PATENT NO.	CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO. A 20050310 JP 2004-199624	DATE
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PI	FR 2868083 FR 2868083 AU 2005232002 CA 2559562 WO 2005098070 W: AE, AG, I CN, CO, O	CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO. A1 20050930 FR 2004-3038 B1 200610721 A1 20051020 AU 2005-232002 A1 20051020 CA 2005-2559562 A2 20051020 WO 2005-FR684 A3 20061005 AL, AM, AT, AU, AZ, BA, BB, BG, BR, BW, BY, ER, CU, CZ, DE, DK, DM, DZ, EC, EE, EG, ES, E	20040324 20050321 20050321 20050321 3Z, CA, CH,

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L24 ANSWER 25 OF 69 USPATFULL on STN
    US 20050265886 A1 20051201
L24 ANSWER 26 OF 69 USPATFULL on STN
    US 20050252580
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L24 ANSWER 27 OF 69 USPATFULL on STN
    US 20050207687
                    A1 20050922
L24 ANSWER 28 OF 69 USPATFULL on STN
    US 20050161125
                    A1 20050728
L24 ANSWER 29 OF 69 USPATFULL on STN
    US 20050058567 A1 20050317
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L24 ANSWER 32 OF 69 CAPLUS COPYRIGHT 2009 ACS on STN
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                                     _____
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    JP 2004292929 A
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L24 ANSWER 33 OF 69 CAPLUS COPYRIGHT 2009 ACS on STN
    PATENT NO. KIND DATE APPLICATION NO. DATE
    JP 2004025205 A
JP 3789856 B2
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20060628
L24 ANSWER 34 OF 69 USPATFULL on STN
    US 20040129243 A1 20040708
     US 7472674
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L24 ANSWER 35 OF 69 USPATFULL on STN
PI US 20040055667 A1 20040325
     US 6911268
                      B2 20050628
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PT

L24	ANSWER 36 OF 69 PATENT NO.	CAPLUS COPYRIGHT KIND DATE	2009 ACS on STN APPLICATION NO.	DATE
PI	JP 2003147482 JP 3851147	A 2003052 B2 2006112	APPLICATION NO. 	20011114
L24	ANSWER 37 OF 69 PATENT NO.	CAPLUS COPYRIGHT KIND DATE	2009 ACS on STN APPLICATION NO. 	DATE
PI	JP 2003147481 JP 3851146	A 2003052 B2 2006112	21 JP 2001-349097	20011114
L24	ANSWER 38 OF 69 PATENT NO.	CAPLUS COPYRIGHT	2009 ACS on STN APPLICATION NO.	DATE
PI	JP 2003147479 JP 3901994	A 2003052 B2 2007040	APPLICATION NO	20011114
L24	ANSWER 39 OF 69 PATENT NO.	CAPLUS COPYRIGHT KIND DATE	2009 ACS on STN APPLICATION NO. 	DATE
PI	JP 2003147436 JP 3750737	A 2003052 B2 2006030	21 JP 2001-349099 01	20011114
L24	ANSWER 40 OF 69 PATENT NO.	CAPLUS COPYRIGHT KIND DATE	2009 ACS on STN APPLICATION NO.	DATE
PI	JP 2003147435 JP 3747365	A 2003052 B2 2006022	APPLICATION NO	20011114
L24	ANSWER 41 OF 69 PATENT NO.	CAPLUS COPYRIGHT	2009 ACS on STN APPLICATION NO.	DATE
PI	JP 2003073735 JP 4213370	A 2003031 B2 2009012	APPLICATION NO. 	20010828
L24	PATENT NO.	KIND DATE	2009 ACS on STN APPLICATION NO.	DATE
PI	FR 2830261	A1 2003040	94 FR 2002-11966	20020927
	JP 2003183770 JP 3468239	A 2003070 B2 2003111	.7 JP 2002-112457	20020415
	CN 1410581	A 2003041	.6 CN 2002-144079	20020930
	US 20030084965 US 6838048	A1 2003050 B2 2005010	14 FR 2002-11966 13 JP 2002-112457 16 CN 2002-144079 15 US 2002-259744	20020930
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	CN 1210432	C 2005071	.3	20030130
	US 20030152476	A1 2003081	.4 US 2003-357419	20030204

	ANSWER 44 OF 69 US 20030178105 US 6764645	A1 20030925	
L24 PI	ANSWER 45 OF 69 US 20030152476	USPATFULL on STN A1 20030814	
L24 PI	ANSWER 46 OF 69 US 20030138343 US 6797231	A1 20030724	
	ANSWER 47 OF 69 US 20030113223 US 6783728	USPATFULL on STN A1 20030619 B2 20040831	
L24 PI	ANSWER 48 OF 69 US 20030084965 US 6838048	USPATFULL on STN A1 20030508 B2 20050104	
L24 PI	ANSWER 49 OF 69 US 20030066577 US 6761853	USPATFULL on STN A1 20030410 B2 20040713	
	ANSWER 50 OF 69 US 6475305	USPATFULL on STN B1 20021105	
L24	ANSWER 51 OF 69 PATENT NO.	CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO. A 20010828 JP 2000-44821	DATE
PI	JP 2001234279	A 20010828 JP 2000-44821	20000222
L24	DATENT NO	CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO.	DATE
PI	JP 2001192765	A 20010717 JP 2000-193780 B2 20020204 A2 20020102 BP 2000-125344	20000628
	R: AT, BE, C	A2 20020102 EP 2000-125344 CH, DE, DK, ES, FR, GB, GR, IT, LI, LU, NL, LT, LV, FI, RO	20001130 SE, MC, PT,
L24	ANSWER 53 OF 69 PATENT NO.	CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO.	DATE
PI		A 20010605 JP 1999-331045	19991122
L24	ANSWER 54 OF 69 PATENT NO.	CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO.	DATE
PI	JP 2001131684	A 20010515 JP 1999-314366	19991104
L24	ANSWER 55 OF 69 PATENT NO.	CAPLUS COPYRIGHT 2009 ACS ON STN KIND DATE APPLICATION NO. A1 20000803 WO 2000-JP369	DATE
PI	WO 2000044953 W: CA, CN, H	A1 20000803 WO 2000-JP369	20000125
		CH, CY, DE, DK, ES, FI, FR, GB, GR, IE, IT,	LU, MC, NL,
	CA 2323952 EP 1069198	A1 20000803 CA 2000-2323952 A1 20010117 EP 2000-900930	20000125
	R: AT, BE, (CH, DE, DK, ES, FR, GB, GR, IT, LI, LU, NL,	SE, MC, PT,
	CN 1113973	C 20030709 CN 2000-800083	20000125

	JP 2000282172 JP 3680674 JP 2001214241 JP 3680708 US 6475305	A 20001010 JP 2000-18101 B2 20050810 A 20010807 JP 2000-204882 B2 20050810 B1 20021105 US 2000-669552	20000127 20000706 20000926
L24	ANSWER 56 OF 69 PATENT NO.	CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO.	DATE
PI	JP 2000273574	A 20001003 JP 1999-81629	19990325
L24	ANSWER 57 OF 69 PATENT NO.	CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO.	DATE
PI	JP 2000219936	A 20000808 JP 1999-23459	19990201
L24		CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO.	DATE
PI	JP 2000026933 JP 3893756	A 20000125 JP 1998-192878 B2 20070314	19980708
L24	PATENT NO.	CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO.	DATE
PI	JP 11001743 JP 3489656	A 19990106 JP 1998-52103 B2 20040126	19980304
L24 PI	ANSWER 60 OF 69 US 5972129	USPATFULL on STN 19991026	
	ANSWER 61 OF 69 US 5922145	USPATFULL on STN 19990713	
L24	ANSWER 62 OF 69 PATENT NO.	CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO.	DATE
PI	JP 10152754 JP 3724142	A 19980609 JP 1997-249242 B2 20051207	19970912
L24		CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO.	DATE
PI	JP 09025541	A 19970128 JP 1995-175318	19950712
L24 PI		USPATFULL on STN 19970708	
L24	ANSWER 65 OF 69 PATENT NO.	CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO.	DATE
ΡI	JP 07150293	A 19950613 JP 1993-296667	19931126
L24		CAPLUS COPYRIGHT 2009 ACS on STN KIND DATE APPLICATION NO.	DATE
ΡI	JP 06248341	A 19940906 JP 1993-33687	19930223
L24 PI		USPATFULL on STN 19920428	
L24	ANSWER 68 OF 69	USPATFULL on STN	

DIT	TIC 4001447	19910211

L24 ANSWER 69 OF 69 USPATFULL on STN PI US 4840073 19890620

=> d cost.		
COST IN U.S. DOLLARS	SINCE FILE	TOTAL
	ENTRY	SESSION
CONNECT CHARGES	15.24	32.59
NETWORK CHARGES	0.77	2.31
SEARCH CHARGES	105.28	582.80
DISPLAY CHARGES	123.01	228.13
FULL ESTIMATED COST	244.30	845.83
DISCOUNT AMOUNTS (FOR QUALIFYING ACCOUNTS)	SINCE FILE	TOTAL
	ENTRY	SESSION
CA SUBSCRIBER PRICE	0.00	-0.82

IN FILE 'CAPLUS, USPATFULL, SCISEARCH' AT 19:58:51 ON 31 JAN 2009

=> logoff ALL L# QUERIES AND ANSWER SETS ARE DELETED AT LOGOFF? (Y)/N/HOLD:y	LOGOFF	
COST IN U.S. DOLLARS	SINCE FILE	TOTAL
	ENTRY	SESSION
FULL ESTIMATED COST	244.30	845.83
DISCOUNT AMOUNTS (FOR QUALIFYING ACCOUNTS)	SINCE FILE	TOTAL
	ENTRY	SESSION
CA SUBSCRIBER PRICE	0.00	-0.82

STN INTERNATIONAL LOGOFF AT 19:59:05 ON 31 JAN 2009